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# Recommended Practice

**Software Reliability** 



**GAIAA** 

### American National Standard

# Recommended Practice for Software Reliability

**Sponsor** 

American Institute of Aeronautics and Astronautics

Approved February 23, 1993

American National Standards Institute

#### **Abstract**

This recommended practice describes an approach to estimating and predicting the reliability of software. It provides information necessary for the application of software reliability measurement to a project, lays a foundation for building consistent methods, and establishes the basic principle for collecting the performance data needed to assess the reliability of software. The document describes how any user may participate in on-going, software reliability assessments or conduct site or package specific studies.

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#### **FOREWORD**

This American National Standard Recommended Practice for Software Reliability has been sponsored by the American Institute of Aeronautics and Astronautics (AIAA) as part of its standards program. It originated within the Space-Based Observation Systems Committee on Standards (SBOS/CoS) and was developed by the Software Reliability Working Group. Members of the working group served voluntarily and without compensation; they are not necessarily members of AIAA. This document represents a consensus of opinions on software reliability measurement from individuals inside and outside AIAA who have expressed an interest in participating in the development of the recommended practice.

Software reliability engineering (SRE) is an emerging discipline. This recommended practice describes an approach to estimating and predicting the reliability of software and is intended to provide a foundation on which practitioners and researchers can build consistent methods. It is intended to meet the needs of software practitioners and users who are confronted with varying terminology for reliability measurement and a plethora of models and data collection methods. This recommended practice contains information necessary for the application of software reliability measurement to a project. It includes guidance on the following:

- Common terminology
- Software reliability estimation procedure
- Model selection
- Data collection procedure for use with the AIAA software reliability database
- Open research questions
- Predicting system failure rates.

This recommended practice was developed to meet the needs of software reliability practitioners and researchers. Practitioners are considered to be the following:

- Managers
- Technical managers and acquisition specialists
- Software engineers
- · Quality and reliability engineers.

Sections 1-4 should be read by all recommended practice users. Section 5 and Appendices E and F provide the basis for establishing the process and the potential uses of the process. Section 7 provides the foundation for establishing a software reliability data collection program, as well as what information needs to be collected to support the recommended models described in Section 6 and Appendix A. Appendix B identifies tools that support the reliability database, the recommended models and the analysis techniques described in Section 5 and Appendices E and F. Finally, to improve the state of the art in software reliability engineering continuously, Appendix D describes research opportunities for consideration. Recommended Practice users typically review Chapters 1-4 and begin applying the techniques described in Sections **5, 6** and 7, concluding with the appendix on reliability tools.

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The viewpoints expressed in this recommended practice are subject to change, depending on developments in the state of the art and comments received from users of the recommended practice. Comments are welcome from any interested party, regardless of membership affiliation with **AIAA**. Comments should be directed to:

AIAA Headquarters Standards Department 370 **L'Enfant** Promenade, SW Washington, DC **20024-25** 18

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At the time this recommended practice was completed, the Software Reliability Working Group had the following members:

David M. Siefert (NCR Corporation),
Working Group Chairman
Ted Keller (IBM Corporation), Vice Chm.
George Stark(Mitre Corporation), Vice Chm.
and Tools Team Chair
William Farr (Naval Surface Warfare Cntr),

Recommended Models Team Chair
Stephen Kelly (Kaman Sciences), Database
Team Chair

Herbert Hecht
Myron Lipow
Michael Lyu
John Musa
Andrea Sebera
Victor Selman
Martin Shooman
Allen Nikora
Norman Schneidewind

Other **SBOS/CoS** Members who participated in the project:

Frank Ackerman Myles R. Berg Charles A. Beswick Ben Bly James A. Boyd Anthony Bukowski John Collins Garrett C. Covington Michael Dewalt Hartwig Dirscherl Janet R. Dunham William Everett George Finelli Dean Garlick Richard Grimaldi S hahid Habib David Hamilton Allen Hankinson Rick Karcich Bernice J. Mays Martha McClure F. A. Patterson-Hine

J. Raja

George Schick V. Devon Smith

Paul F. Uhlir

Richard D. Stutzke

Stephanie White Ken. S. Williamson Frank Y. Yap

The following individuals have contributed to the review and input of this document:

Bev Littlewood Harvey Fiala J. Rayon Lisa Brownsword Roger Martin Robert Tausworthe

#### **Supporting Organizations:**

Aerospace American University AT&T Corporation Ball Aerospace Bendix Computer Sciences Corporation Embassy of India General Dynamics Corporation Grumman Corporation Hughes Aircraft IBM Corporation John Hopkins Univ. Applied Physics Lab. Jet Propulsion Laboratory Kaman Sciences/DACS Lockheed Company Loral Space Systems MITRE Corporation NASA National Institute for Standards & Technology Naval Postgraduate School Naval Surface Warfare Center **Orien** Polytechnic University Rome Air Develoment Center Research Triangle Institute **Science** Applications International Corp. SoHaR Storage Technology Corporation. TRW UNISYS USSC/ANS

The AIAA **SBOS/CoS** (Andrea F. Sebera, Chair) approved the document in May 1992.

The AIAA Standards Technical Council (William. W. Vaughan, Chairman) approved the document in November 1992.

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#### 1.0 INTRODUCTION

#### 1.1 Scope

Software Reliability Engineering (SRE) is an emerging discipline. SRE is the application of statistical techniques to data collected during system development and operation to specify, predict, estimate, and assess the reliability of software-based systems. This recommended practice defines a practical methodology for software reliability engineering.

The Recommended Practice for Software Reliability provides a foundation for practitioners and researchers. It supports the need of software practitioners who are confronted with inconsistent methods and varying terminology for reliability estimation and prediction, as well as a plethora of models and data collection methods. It supports researchers by defining common terms, by identifying criteria for model comparison, and by identifying open research problems in the field

This document provides guidance on the following:

- Common terminology
- Software reliability estimation and procedure
- Model selection
- Data collection procedure for use with the AIAA software reliability database

This recommended practice is applicable to in-house, commercial, and third-party **soft-**

ware projects. It has been developed to support a systems reliability approach. As illustrated in Figure 1, the AIAA Software Reliability Engineering Recommended Practice considers hardware and ultimately systems characteristics.

#### 1.2 Purpose

The AIAA Recommended Practice for Software Reliability is intended to be used from the start of the integration test phase through the operational use phase of the software life cycle. It also provides input to the planning process for reliability management. It is assumed that the use of this handbook has been preceded by an identification and analysis of user requirements.

The Recommended Practice describes activities and qualities of a software reliability estimation and prediction program. It describes a framework that permits assessment of risk and prediction of failure rates, recommends a set of models for software reliability estimation and prediction, and specifies mandatory as well as recommended data collection requirements.

#### 1.3 Intended Audience and Benefits

The Recommended Practice is intended for use by both practitioners (e.g., software developers, software acquisition personnel, technical managers, and quality and reliability personnel) and researchers. Its purpose is to provide both practitioners and researchers with a common baseline for discussion and to define a procedure for assessing the reliability of software, It is assumed that users of this

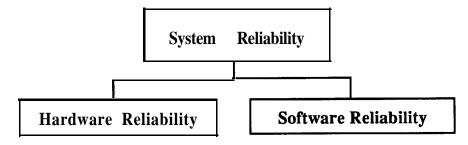


Figure 1 System Reliability Characteristics

recommended practice have a basic understanding of the software life cycle and an understanding of statistical concepts.

This recommended practice is intended to be used in support of designing, developing and testing software. This includes software quality and software reliability activities. It also serves as a reference for research on the subject of software reliability.

### 1.4 Applications of Software Reliability Engineering

The techniques and methodologies presented in this handbook have been successfully applied to software projects by industry practitioners in order to do the following:

- Determine whether a specific software process is likely to produce code which satisfies a given software reliability requirement,
- Determine the size of a software maintenance effort by predicting the failure rate during the operational phase,
- Provide a metric for process improvement evaluation,
- Assist software safety certification,
- Determine when to release a software system, or to stop testing it,
- Estimate the occurrence of the next failure for a software system,
- Identify elements in a software system which are leading candidates for re-design to improve reliability,
- Measure reliability of a software system in operation, using this information to control change to the system.

It is the intent of this recommended practice to enable other software practitioners to make similar determinations for their particular software systems, as needed. Special attention should be given in the application of these practices to avoid violation of the assumptions inherent in each modeling

technique. Data acquisition procedures and model selection criteria are provided and discussed in order to assist in these efforts.

### 1.5 Relationship to Hardware Reliability

The creation of software and hardware products are alike in many ways. and can be similarly managed throughout design and development. While the management techniques may be similar, there are genuine differences between hardware and software [LIPO86, KLIN80] . For example:

- Changes to hardware require a series of important and time-consuming steps: capital equipment acquisition, component procurement, fabrication, assembly, inspection, test and documentation. Changing software is frequently more feasible (although effects of the changes are not always clear) and oftentimes requires only testing and documentation.
- Software has no physical existence. It includes data as well as logic. Any item in a file can be a source of failure.
- Software does not wear out. Furthermore, failures attributable to software faults come without advance warning and often provide no indication they have occurred. Hardware, on the other hand, often provides a period of graceful degradation.
- Software may be more complex than hardware, although exact software copies can be produced, whereas manufacturing limitations affect hardware.
- Repair generally restores hardware to its previous state. Correction of a software fault always changes the software to a new state.
- Redundancy and fault tolerance for hardware are common practice. These concepts are only beginning to be practiced in software.
- Software developments have traditionally made little use of existing components. Hardware is manufactured with standard

parts.

- Hardware reliability is expressed in wall clock time. Software reliability is expressed in execution time.
- A high rate of software change can be detrimental to software reliability.

Despite the above differences, hardware and software reliability must be managed as an integrated system attribute. However, these differences must be acknowledged and accommodated by the techniques applied to each of these two types of subsystems in reliability analyses.

#### 2.0 TERMINOLOGY

This chapter defines terms that are commonly used throughout the recommended practice. The bases for most definitions are from the ANSI / IEEE Standard Glossary of Software Engineering Terminology, STD-729-1991.

**Calendar time -** Chronological time, including time during which a computer may not be running.

**Clock time** - Elapsed wall clock time from the start of program execution to the end of program execution.

Error • (1) A discrepancy between a computed, observed or measured value or condition and the true, specified or theoretically correct value or condition. (2) Human action that results in software containing a fault. Examples include omission or misinterpretation of user requirements in a software specification, and incorrect translation or omission of a requirement in the design specification. This is not a preferred usage.

**Execution time** • (1) The amount of actual or central processor time used in executing a program. (2) The period of time during which a program is executing.

**Failure** - (1) The inability of a system or system component to perform a required function within specified limits. A failure may be produced when a fault is encountered and a loss of the expected service to the user

results. (2) The termination of the ability of a functional unit to perform its required function. (3) **A** departure of program operation from program requirements.

Failure rate - (1) The ratio of the number of failures of a given category or severity to a given period of time; for example, failures per second of execution time, failures per month. Synonymous with failure intensity. (2) The ratio of the number of failures to a given unit of measure; for example, failures per unit of time, failures per number of transactions, failures per number of computer runs.

**Failure Severity -** A rating system for the impact of every recognized credible software failure mode. For example,

- Severity #1 Loss of life or system
- Severity #2 Affects ability to complete mission objectives
- Severity #3 Workaround available, therefore minimal effects on procedures (mission objectives met)
- Severity #4 Insignificant violation of requirements or standards, not visible to user in operational use
- Severity #5 Cosmetic issue which should be addressed or tracked for future action, but not necessarily a present problem.

**Fault** • (1) A defect in the code that can be the cause of one or more failures. (2) An accidental condition that causes a functional unit to fail to perform its required function. Synonymous with bug.

**Fault Tolerance -** The survival attribute of a system that allows it to deliver the required service after faults have manifested themselves within the system.

**Firmware** • (1) Computer programs and data loaded in a class of memory that cannot be dynamically modified by the computer during processing. (2) Hardware that contains a computer program and data that cannot be changed in its user environment. The

computer programs and data contained in firmware are classified as software; the circuit containing the computer program and data is classified as hardware. (3) Program instructions stored in a read-only storage. (4) An assembly composed of a hardware unit and a computer program integrated to form a functional entity whose configuration cannot be altered during normal operation. The computer program is stored in the hardware unit as an integrated circuit with a fixed logic configuration that will satisfy a specific application or operational requirement.

**Integration** - The process of combining software elements, hardware elements or both into an overall system

**Maximum Likelihood Estimation -** A form of parameter estimation in which selected parameters maximize the probability that observed data could have occurred.

Module • (1) A program unit that is discrete and identifiable with respect to compiling, combining with other units and loading; for example, input to or output from an assembler, compiler, linkage editor or executive routine. (2) A logically separable part of a program.

**Operational** - Pertaining to the status given a software product once it has entered the operation and maintenance phase.

**Parameter** • A variable or arbitrary constant appearing in a mathematical expression, each value of which restricts or determines the specific form of the expression.

**Quality** • The totality of features and characteristics of a product or service that bears on its ability to satisfy given needs.

**Subsystem -** A group of assemblies, components or both combined to perform a single function.

**Software Quality** • (1) The totality of features and characteristics of a software product that bear on its ability to satisfy given needs; for example, to conform to specifications. (2) The degree to which software possesses a desired combination of attributes. (3) The

degree to which a customer or user perceives that software meets his or her composite expectations. (4) The composite characteristics of software that determine the degree to which the software in use will meet the expectations of the customer.

**Software Reliability** • (1) The probability that software will not cause the failure of a system for a specified time under specified conditions. The probability is a function of the inputs to and use of the system, as well as a function of the existence of faults in the software. The inputs to the system determine whether existing faults, if any, are encountered. (2) The ability of a program to perform a required function under stated conditions for a stated period of time.

**Software Reliability Engineering** • the application of statistical techniques to data collected during system development and operation to specify, predict, estimate, and assess the reliability of software-based systems.

**Software Reliability Estimation - The** application of statistical techniques to observed failure data collected during system testing and operation to assess the reliability of the software.

**Software Reliability Model** - A mathematical expression that specifies the general form of the software failure process as a function of factors such as fault introduction, fault removal and the operational environment.

**Software Reliability Prediction - A** forecast of the reliability of the software based on parameters associated with the **software** product and its development environment.

**System** • (1) A collection of people, machines and methods organized to accomplish a set of specific functions. (2) An integrated whole that is composed of diverse, interacting, specialized structures and subfunctions. (3) A group or subsystem united by some interaction or interdependence, performing many duties but functioning as a single unit.

### 3.0 REFERENCE DOCUMENTS

This section contains reference documents that are applicable to the field of software reliability engineering.

#### 3.1 Primary Documents

The following list of standards should be reviewed prior to implementation of this hand-book:

- ANSI / IEEE Std 729-1991, "IEEE Standard Glossary of Software Engineering Terminology"
- MIL-Std 756, "Reliability Modeling and Prediction"

#### 3.2 Other Documents

The following list of documents provide additional information applicable to the scope of the handbook.

- IEEE Std 982.1-1988, "IEEE Standard Dictionary of Measures to Produce Reliable Software"
- IEEE Std 982.2-1988, "IEEE Guide for the Use of IEEE Standard Dictionary of Measures to Produce Reliable Software"
- IEEE Std 1061-1992, "IEEE Standard for a Software Quality Metrics Methodology"
- MIL-Std-785, "Reliability Programs for Systems and Equipment"
- IEEE Std 1074, "Standard for Life-cycle Processes"
- MIL-HDBK 217, "Reliability Prediction of Electronic Equipment"

#### 4.0 SOFTWARE RELIABILITY MODELING • OVERVIEW, CONCEPTS, AND ADVANTAGES

Software is a complex intellectual product. Inevitably, some errors are made during requirements formulation as well as during designing, coding and testing the product. The development process for high-quality software includes measures that are intended to discover and correct faults resulting from these errors, including reviews, audits, screening by language-dependent tools and several levels of test. Managing these errors involves describing, classifying and modeling the effects of the remaining faults in the delivered product and thereby helping to reduce their number and criticality.

Dealing with faults costs money. It also impacts development schedules and system performance (through increased use of computer resources such as memory, CPU time and peripherals requirements). As is usually recognized, there can be too much as well as too little effort spent dealing with faults. Thus the system engineer (along with management) can use reliability estimation and prediction to understand the current status of the system and make tradeoff decisions.

This section describes the basic concepts involved in software reliability engineering and addresses the advantages and limitations of software reliability prediction and estimation.

#### 4.1 Basic Concepts

There are at least two significant differences between hardware reliability and software reliability. First, software does not fatigue, wear out or burn out. Second, due to the accessibility of software instructions within computer memories, any line of code can contain a fault that, upon execution, is capable of producing a failure.

A software reliability model specifies the general form of the dependence of the failure process on the principal factors that affect it: fault introduction, fault removal and the operational environment.

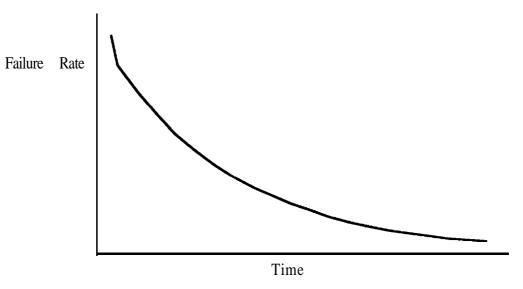


Figure 2 Software Reliability Measurement Curve

The failure rate (failures per unit time) of a software system is generally decreasing due to fault identification and removal. At a particular time, it is possible to observe a history of the failure rate of the software. Software reliability modeling is done to estimate the form of the curve of the failure rate by statistically estimating the parameters associated with the selected model. The purpose of this measure is two-fold: (1) to estimate the extra execution time required to meet a specified reliability objective and (2) to identify the expected reliability of the software when the product is released. This procedure is important for cost estimation, resource planning, schedule validation and quality prediction for software maintenance management.

### **4.2 Limitations of Software** Reliability Prediction and Estimation

There are two types of models that can be applied for software reliability measurement. First, there are prediction models which make use of parameters associated with the software product and its development environment to predict the reliability of a software product. Second, there are estimation models which apply statistical techniques to the observed failures during software testing and operation to forecast the product's reliability. This section describes the limitations of each type of model.

Both prediction and estimation models need good data if they are to yield good forecasts. Good data implies accuracy (that data is truthfully recorded at the time the events occurred) and pertinence (that data relates to an environment that is equivalent to the environment for which the forecast is to be valid). A negative example with respect to accuracy is the restricting of failure report counts to those which are completely filled out. This is negative because they may represent a biased sample of the total reports. A negative example with respect to pertinence would be the use of data from early test runs at an uncontrolled workload to forecast the results of a later test executed under a highly controlled workload.

### **4.2.1** Prediction Model Advantages / Limitations

In prediction models, the failure probability of a program in development is forecast by comparing it to the known failure probability (or other reliability parameters) of an existing program. The existing program is known as a *proof* program. The advantage of this procedure is that it can be performed at any time during the development, whereas reliability estimation depends on the availability of operational or test data. The validity of the prediction depends on (a) the degree of similarity between the program under development and the proof program

(for which failure rates are known), and (b) the quality of known failure rate data.

When there is direct equivalence between the proof program and the program under development, reliability prediction is a specific application of the Similar Item Method as defined in MIL-STD-756B. The criteria established in MIL-STD-756B for application of this method include:

- Design similarity
- . Similarity of service use profile
- Procurement and project similarity
- · Proof of reliability achievement.

Because all these criteria can be met only under rare circumstances, alternative methods are usually followed. The most applicable alternative for software involves the following steps:

- (1) Estimate the size of the source code. This is a routine step in software development. Many organizations have a size growth model that compensates for the usual underestimating of program size during early stages of development.
- (2) Estimate the fault density (faults per line of source code) at the start of formal test (a test activity applicable to the program as a whole and for which computer usage hours will be collected). The preferred approach is to use the fault density determined for a similar program created in the same environment. Where this is not possible, a fault density ranging from 0.001 (for programs developed in a highly disciplined environment and by programmers that have extensive background in the specific application) to 0.01 in a more routine environment may be assumed [MUSA87, Table 5.21.
- 3) The product of (1) and (2), gives the expected number of faults in the code at the start of formal test. This number corresponds to ω<sub>0</sub> in the Musa Basic

Model [MUSA87, Eq. 5.2] and to N in the Jelinski-Moranda Model (Appendix A).

In some environments, the relation between the failure rate at a given point in the development and the fault content at the start of test may be known from experience. In that case, the local factor should be used and the following steps can be omitted.

- (4) The key considerations for most models are: (a) the initial number of faults, (b) the probability of executing a specific fault during a single execution (the fault exposure ratio), and (c) the time for which the prediction is to be valid. The latter consideration is at the user's discretion; in some models the time is defined in terms of the number of faults that have been found. The value of the fault exposure ratio is  $4.0 \pm 2x10-7$  for 8 out of 13 examples shown in [MUSA87, Table 5.61; the total range is  $1.41 \times 10^{-7}$ to 10.6x10-7. Where the fault exposure ratio for similar programs is known, that value should be used in preference to the default values of the previous sentence.
- (5) The failure probability per fault and unit time is denoted by  $\phi$  in the Jelinski-Moranda Model and by fK in the Musa Basic Model. The factor f is the frequency at which a given (object) instruction will be accessed by the program. It can be computed from f =r/I, where r is the execution rate of the computer and I is the number of object instructions in the program. dimension of r is instructions per unit time and the time units must be consistent with those for which the failure rate prediction is to be generated. Since execution rate is normally expressed per second and failure rates are expressed per hour, an appropriate conversion has to be performed.
- (6) The initial failure rate can then be predicted as  $\lambda_0 = fK\omega_0$  for the Musa Basic Model. With these parameters, the expected failure rate at a future point in

time (or after a given number of faults have been detected) can be found by using most of the models described in Section 6 or Appendix A.

Other prediction models use data on the application area, development and test environments, and characteristics of the code (e.g., complexity, modularity) [MCCA87]. These are alternative ways of estimating the fault density and / or the fault exposure ratio. To date, none of these approaches has been shown to be widely applicable. Their use should be restricted to environments where their validity has been demonstrated

### **4.2.2** Estimation Model Advantages / Limitations

The premise of most estimation models is that the failure rate is a direct function of the number of faults in the program and that the failure rate will be reduced (reliability will be increased) as faults are detected and eliminated during test or operations. This premise is reasonable for the typical test environment and it has been shown to give credible results when correctly applied. However, the results of estimation models will be adversely affected by:

- · Change in failure criteria
- Significant changes in the code under test
- Significant changes in the computing environment.

All of these factors will require a new set of reliability model parameters to be computed. Until these can be established, the effectiveness of the model will be impaired. Estimation of new parameters depends on the measurement of several execution time intervals between failures.

Major changes can occur with respect to several of the above factors when software becomes operational. In the operational environment, the failure rate is a function of the fault content of the program, of the variability of input and computer states, and of software maintenance policies. The latter two factors are under management control and are **fre**-

quently utilized to achieve an expected or desired range of values for the failure rate or the downtime due to software causes. Examples of management action that decrease the failure rate include: avoidance of high work loads and avoidance of data combinations that have caused previous failures [GIFF84,IYER83]. Software in the operational environment may not exhibit the reduction in failure rate with execution time that is an implicit assumption in most estimation models [HECH86a]. Knowledge of the management policies is therefore essential for selection of a software reliability estimation procedure for the operational environment. Thus, the estimation of operational reliability from data obtained during test may not hold true during operations.

Another limitation of software reliability estimation models is their use for verifying ultra-high requirements. For example, if a program executes successfully for x hours, there is maybe a 0.5 probability that it will survive the next x hours without failing [LITT90]. Thus, to have the kind of confidence needed to verify a 10-9 requirement would require that the **software** execute failure-free for several billion hours. Clearly, even if the software had achieved such a reliability, one could never assure that the requirement was met. The most reasonable verifiable requirement is somewhere in the 10-3 or 10-4 range.

It is important to understand the nature of the program when discussing ultra-high requirements. Many ultra-reliable applications are implemented on relatively small, slow, inexpensive computers. Furthermore, the critical programs are small (less than 1000 source lines of code) and execute infrequently during an actual mission. With this knowledge, it may be feasible to test the critical program segment on several faster machines, considerably reducing the required test time.

Furthermore, where very high reliability requirements are stated (failure probabilities < 10<sup>-6</sup>) they frequently are applicable to a software controlled process together with its protective and mitigating facilities and therefore they tend to be overstated if applicable to

the process alone. An example of a protective facility is an automatic cut-off system for the primary process and reversion to analog or manual control. An example of a mitigation facility is an automatic sprinkler system that significantly reduces the probability of file damage in case the software controlled process generates excessive heat. If the basic requirement is that the probability of extensive fiie damage shall not exceed 10-6 per day, and if both protecting and mitigating facilities are in place, it is quite likely that further analysis will show the maximum allowable failure rate for the software controlled process to be on the order of 10-3 per day and hence within the range of current reliability estimation methods.

Where the requirements for the software controlled process proper still exceed the capabilities of the estimation methodology after allowing for protective and mitigating facilities, fault tolerance techniques may be applied. These may involve fault tolerance [HECH86b] or functional diversity. An example of the latter is to control both temperature and pressure of steam generation, such that neither one of them can exceed safety criteria. The reduction in failure probability that can be achieved by software fault tolerance depends in a large measure on the independence of failure mechanisms for the diverse implementations. It is generally easier to demonstrate the independence of two diverse functions than it is to demonstrate the independence of two computer programs, and hence functional diversity is frequently preferred.

# 5.0 SOFTWARE RELIABILITY ESTIMATION PROCEDURE

This section provides guidance to the practitioner on how to do software reliability estimation and what types of analysis can be performed using the technique. It defines a generic step-by-step procedure for executing software reliability estimation and describes possible analysis using the results of the estimation procedure.

#### 5.1 Generic Procedure

An eleven-step generic procedure for estimating software reliability is listed below. This generic procedure should be tailored to the project and the current life-cycle phase. Some steps will not be used in some applications, but the structure provides a convenient and easily remembered standard approach. The following steps can be used to generate a **checklist** for reliability programs:

- 1) Identify Application
- 2) Specify the Requirement
- 3) Allocate the Requirement

Since this document is concerned only with the test through operational life-cycle activities, only steps (4) through (11) are discussed.

#### 4) Define Failure

- 5) Characterize the Operational Environment
- 6) Select Tests
- 7) Select Modes
- 8) Collect Data
- 9) Estimate Parameters
- 10) Validate the Model
- 11) Perform Analysis

#### **5.1.1 Define Failure**

A *project specific* failure definition is usually negotiated by the testers, developers, and users. It is agreed upon prior to the beginning of test. In spite of this necessary tailoring, there are often commonalities in the definition among similar products (e.g., most people agree that a software bug that when encountered stops all processing is a failure). The important consideration is that the definition be consistent over the life of the project.

There are a number of specific considerations

relating to the interpretation of failure for systems. The analyst must determine the answers to these questions:

- Is a failure counted if it is consciously decided not to seek out and remove the cause of a particular failure?
- Are repeated failures counted?
- What is a failure in a fault-tolerant system?
- Are a series of failures counted if they are triggered by data degradation?

A discussion of each of these considerations is provided in [MUSA87, pp 77-85].

Projects need to classify failures by their severity. An example classification is provided in Section 2. Classes are usually separated by an order of magnitude in costs. Impact can not ordinarily be estimated with great precision. It is desirable to consider severity by type, and by user requirement.

For some projects, there appears to be a relative homogeneity with respect to time-of-failure among high-severity failures. For example, if 10 percent of the failures occurring early in test fall in a particular class, about the same percentage will be expected to be found in that class late in test. This permits making, for example, statistical estimates based on all data to achieve a smaller confidence interval and then adjusting them to **per class** estimates. It also is possible to weight failure data by a variable (such as cost) associated with class and to obtain compound estimates such as failure cost rather **than** failure intensities.

It is recommended that failure times be recorded in execution time. However, should execution time not be readily available, elapsed clock time is a satisfactory approximation if machine utilization is constant (when averaged over a time period comparable to the times between failures). If utilization is not constant, one often can weight the clock time by a measure that is proportional to the utilization, such as number of uses of a real-time system. Execution time also can be approximated by natural

units like transactions.

When failure times are collected from multiple machines functioning simultaneously, intervals between failures should be counted by considering execution time on all machines. If the machines have different average instruction execution rates, execution times should be adjusted to a reference machine [MUSA87, pp 162-165].

### **5.1.2** Characterize the Operational Environment

Characterization of the operational environment has three aspects: 1) system configuration, 2) system evolution, and 3) system operational **profile**.

System configuration is the arrangement of the system's components. Software-based systems are just that; they can not be pure but must include hardware as well as software components.

Distributed systems are a type of system configuration. The purpose of determining the system configuration is twofold:

- To determine how to allocate system reliability to component reliabilities
- To determine how to combine component reliabilities to establish system reliability [MUSA87, pp 85-106].

In modeling software reliability, it is **necessary** to recognize that systems frequently evolve as they are tested. That is, new code or even new components are added. Special techniques for dealing with evolution are provided in [MUSA87, pp 166-176].

The system operational profile characterizes in quantitative fashion how the software will be used. It lists all operations realized by the software and the probability of occurrence and criticality of each operation.

A system may have multiple operational profiles or operating modes. They usually represent difference in function associated with significant environmental variables. For example, a space vehicle may have ascent, **on-**

orbit and descent operating modes. Operating modes may be related to time, installation location, customer or market segment. Reliability can be tracked separately for different modes if they are significant. The only limitation is the extra data collection and cost involved.

#### 5. 1. 3 Select Tests

Many applications of software reliability engineering involve the execution of operations and collection of failure data. Operations should be picked to reflect how the system will actually be used. Reference Appendix C for information that may be useful in determining failure rates. In other words, the test operational profile should represent the field operational profile.

The tester selects one of the following approaches:

- Test duplicates actual operational environments (as closely as possible)
- Testing conducted under more severe conditions; for extended periods of time resulting in failures being accumulated in less than expected time.

The modeling effort must take into account the specific approach taken by the test team to expose faults so that accurate forecasts can be made.

#### 5.1.4 Select Models

The models described in Section 6 have been identified for giving good results in specific environments, but it can not be guaranteed that they will be suitable in new environments. Therefore it is recommended that each user compare several models prior to final selection.

A list of the model selection criteria described in Section 6.1 is provided below:

- Predictive Validity
- Ease of Parameter Measurement
- Quality of Assumptions

- Capability
- Applicability
- Simplicity
- Insensitivity to Noise

In general, each model should be evaluated by these criteria, using the best model to make forecasts.

#### 5.1.5 Collect Data

Data collection must be geared toward the overall objectives of the software reliability effort, such as the attainment of a failure-free interval.

In considering setting up a reliability program, one must avoid several pitfalls. The first is that every bit of information about the program and what happens to it as it evolves over the life cycle needs to be kept. The second is that clearly defined objectives for the data collection process are not necessary. These two pitfalls result in too much effort expended with too little payback. When a massive amount of data is required, it is usually the program manager's people that are impacted. Cost and schedule suffer.

Two additional points that should be kept in mind while planning to collect data and collecting data are: (1) motivate the data collectors, and (2) review the collected data promptly. If these two things are not done, quality will suffer.

A list of the data collection steps detailed in Section 7.1 is provided below:

- 1) Establish the objectives.
- 2) Set up a plan for the data collection process.
- 3) Apply tools.
- 4) Provide training.
- Perform trial run.
- 6) Implement the plan.

- 7) Monitor data collection.
- 8) Evaluate the data as the process continues.
- 9) Provide feedback to all parties.

In general, a process should be established addressing each of these steps, and a successful software reliability data collection program will emerge.

#### **5.1.6** Estimate Parameters

There are three common methods of parameter estimation: method of moments, least squares, and maximum likelihood. Each of these methods has attributes that make it useful. However, maximum likelihood estimation is the most commonly used approach. A full treatment of parameter estimation is provided in [MUSA87, FARR83, and SHOO83]. All of the software reliability engineering tools described in Appendix B perform parameter estimation as one of their capabilities using one or more of these methods.

#### 5.1.7 Validate the Model

Several considerations are involved in properly validating a model for use on a given production project. First, it is necessary to deal with the assumptions of the model under evaluation. Choosing appropriate failure data items and relating specific failures to particular intervals of the life-cycle or change increments often facilitate this task [SCHN92]. Depending on the progress of the *production* project, the model validation data source should be selected from the following, listed in the order of preference:

- 1) **Production** project failure history (if project has progressed sufficiently to produce failures).
- Prototype project employing similar products and processes as the *production* project.
- 3) Prior project employing similar products and processes as the **production** project (reference Appendix C)

Using one of these data sources, the analyst should execute the model for several specific times within the failure history period and then compare the model output to the actual subsequent failure history using one of the following:

- 1) Predictive validity criteria (Section 6.1.1).
- 2) A traditional statistical goodness-of-fit test (e.g., Chi-square or **Kolmogorov-Smirnov**).

It is important that a model be continuously re-checked for validation, even after selection and application, to ensure that the fit to the observed failure history is satisfactory. In the event that a degraded model fit is experienced, alternate candidate models should be evaluated using the procedure above.

#### 5.1.8 Perform Analysis

Once the data has been collected and the model parameters estimated, the analyst is ready to perform the appropriate analysis. This analysis may be to estimate the current reliability of the software, forecast the number of faults remaining in the code, or forecasting a testing completion date. Section 5.2 details a set of common analyses conducted using software reliability theory.

One pitfall to **be** careful of is the combination of a software reliability value into a system reliability calculation. If the analysis calls for producing a system reliability figure and the software reliability is calculated in terms of execution time, it must be converted to calendar time for combination with hardware reliabilities to calculate the system value.

#### 5.2 Recommended Analysis Practice

This section provides details of analysis procedures for some common engineering or management activities that can be aided by software reliability engineering technology. These details are in most cases a description of the analysis that must be performed as the last step of the generic procedure described in Section 5.1. Although this list is far from complete, it is a set to start from.

#### 5.2.1 Estimate Current Reliability

Since software will not fail until the software is executed and a software fault is manifested by the computer, the time measurement based on CPU time for failure data collection is preferred. However, there are approximating techniques if the direct measurement of CPU time is not available [MUSA87, pp 156-158]. When combined with hardware reliability measurement (to form the system reliability prediction) the CPU time also can be transformed to calendar time [MUSA87, pp 113-1391.

Reliability estimations in test and operational phases basically follow the same procedure. However, there is a difference. During the testing phase, software faults are intended to be removed as soon as the corresponding software failures are detected. As a result, the reliability growth could be observed. However, in the operational phase, correcting a software fault involves changes of multiple software copies in the customers' sites, which, unless the failure is catastrophic, is not always done until the next software release.

Therefore, the software failure rate usually

remains constant until a new version is released, in which case a jump in reliability should be observed. Nevertheless, the users might change the use of the software to avoid triggering the known failure. In other words, the operational profile is changed and certain growth of reliability could still be observed.

### **5.2.2 Forecast Achievement of a** Reliability Goal

The date at which a given reliability goal can be achieved is obtainable **from** the software reliability modeling process illustrated in Figure 3. As achievement of the reliability target approaches, the adherence of the actual data to the model should be reviewed and the model calibrated if necessary. Refer to Appendix F, "Using Reliability Models for Developing Test Strategies."

### **5.2.3 Forecast Additional Test Duration**

Additional test duration may be predicted if the initial and objective failure intensities and the parameters of the model are known. (These are identified for each model in Section 6.) For the Musa Basic exponential model we have:

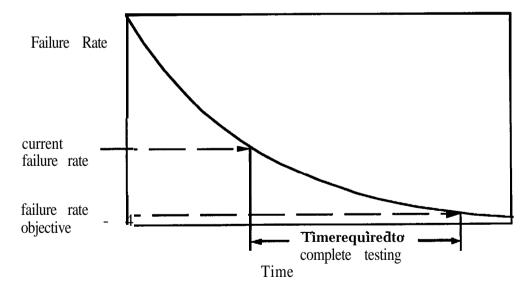


Figure 3 Example Software Reliability Measurement Application

$$At = \frac{v_0}{\lambda_0} ln \left( \frac{\lambda_0}{\lambda_F} \right)$$

where At is the test duration in CPU hr,  $\nu_0$  is the total failures parameter of the model,  $\lambda_0$  is the initial failure intensity, and  $\lambda_F$  is the objective failure intensity.

The formula for the Musa-Okumoto Logarithmic Poisson model is

$$\Delta t = \frac{1}{\theta} \left( \frac{1}{\lambda_F} - \frac{1}{\lambda_0} \right)$$

where  $\theta$  is the failure intensity decay parameter.

Calendar time test duration could be **com**pu ted manually. However, all calculations are generally available in software reliability tools (See Appendix B), and the formulas given above are only occasionally applied manually.

### **5.2.4 Establish Conformance with Acceptance Criteria**

If reliability-related criteria are part of software acceptance, the model should be selected so that its results can be easily interpreted for conformance with the selection criteria. For example, time-to-failure models are consistent with requirements for **failure**free intervals. Failure count models are suitable for establishing conformance with maximum failure rate requirements.

### 5.2.5 Manage Introduction of New Features into Operational Software

Decisions about whether and when to introduce new features into operational software must be made. Introduction of new features carries the risk of adding new faults and hence increasing the failure intensity. This could raise the failure intensity to such a level that the impact on service is unacceptable. Software reliability engineering provides a quantitative way of measuring service and hence a guide for permitting or delaying the

introduction of the new features. Failures are regularly recorded during operation and failure data is entered in a software reliability estimation program, which is run at regular intervals (frequently weekly). A running plot of failure intensity is generated.

A failure intensity objective is selected based on a balance between the need for new features and the need that old features operate reasonably reliably. The proximity of the actual failure intensity to the objective is now used as the criterion for accepting new features. New features are accepted only when the actual failure intensity is sufficiently below the objective that it appears unlikely that the addition of the new features will increase the failure intensity substantially above the objective.

## 5.2.6 Evaluate Reliability Impact of Software Engineering Technology Variables

It is important to know the impact of technology on software reliability. This knowledge will make it possible to design an efficient development process for a particular software product. These impact studies have not been performed to any extent at the writing of this document, but they could and should be. For example, the relationship between effort devoted to design inspection per thousand source lines of code and the change in failure intensity should be studied. This is done by holding other variables constant as design inspection effort is varied. The resultant quantity that is measured could be initial failure intensity at the start of system test.

### **5.2.7 Estimate Maintenance Staffing Requirements**

Three quantities are needed to estimate the staff required to restore systems after a soft-ware failure: first, the average time required for a repair person to restore the system after a failure (including travel time); second, the expected operating time of the software in time units; and third, the expected failure rate of the software in operation.

Multiplying the failure rate by the operational

time yields the expected number of failures per time unit. Using this number and the average time to restore the system, the number of repair personnel can be derived. It is important to assign more repair personnel than this estimate to account for variations in failure occurrence that may result in lower system availability.

#### 5.2.8 Assist Safety Certification

A software safety failure can be defined as any software system behavior that involves risk to human life, risk of injury or risk of equipment damage. Thus, Failure Severity #1 (see Terminology Section 2 - "Failure Severity") failures fall into this category. The failure rate based on the failures in this category can be determined to support software safety certification. It is important to note that reliability is a necessary, although not sufficient, condition to ensure safety and should not be used as the only criterion for safety certification.

#### 6.0 SOFTWARE RELIABILITY ESTIMATION MODELS

There are many ways to develop a software reliability model: (a) describe it as a stochastic process, (b) relate it to a Markov model, (c) define the probability density or distribution function, or (d) specify the hazard function. These approaches are all equivalent and equally correct. There are three general classes of software reliability estimation models: Exponential non-homogeneous Poisson process (NHPP) models, Non-exponential NHPP models and Bayesian models. The following paragraphs describe the characteristics of each general class.

#### **Exponential NHPP Models**

Exponential NHPP models use the stochastic process and the hazard function approach. The hazard function, z(t), is generally a function of the operational time, t. Several different derivations of z(t) are given in [SHOO90a]. The probability of success as a function of time is the reliability function, R(t), which is given by:

$$R(t) = \exp\left[-\int_{0}^{t} z(y) dy\right]$$

Sometimes reliability is expressed in terms of a single parameter: mean time to failure, (MTTF). MTTF is given by:

$$MTTF = \int_{0}^{\infty} R(t) dt$$

On occasion the reliability function may be of such a form that MTTF is not defined. The hazard function (or failure intensity, [MUSA87, pp. 11, 18]) or the reliability function can be used in this case. The hazard function can be constant or can change with time.

Representative models selected for this class include: Shooman's model; Musa's Basic model; Jelinski and Moranda's model (described in Appendix A); and the generalized exponential model (described in Section 6.2). Model objectives, assumptions, parameter estimates, and considerations for using the model are described in the appropriate section.

#### **Non-Exponential NHPP Models**

Non-Exponential NHPP models also use the stochastic process and the hazard function approach. They are generally applicable when testing is done, according to an operational profile that is not uniform in nature. Early fault corrections have a larger impact on the failure intensity function than later ones.

Representative models selected for this class include: Duane's model; Brooks and Motley's Binomial and Poisson models; Yamada's S-shaped model (all described in Appendix A); and Musa and Okumoto's Logarithmic Poisson (described in section 6.2). The assumptions and format of the respective model, its estimates for model fitting, and finally considerations for the employment of the model are described in the appropriate section.

#### **Bayesian Models**

Bayesian models differ from NHPP models in two ways. First, where NHPP models only allow for change in reliability when a fault is discovered and corrected, Bayesian models allow reliability to change based on the length of failure-free testing time periods. Second, NHPP models assume that the hazard function is directly proportional to the number of faults in the program and hence the reliability is a function of this fault count. The Bayesian approach argues that a program can have many faults in unused sections of the code and exhibit a higher reliability than software with only one bug in a frequently exercised section of code. Representative models of this class are those developed by Littlewood [LITT79].

#### 6.1 Criteria for Model Evaluation

This following criteria should be used for conducting an evaluation of software reliability models in support of a given project.

- Model predictive validity: the performance and correctness of the forecast quality of each model. Measures defined for this are: accuracy, trend, bias, and noise.
- Ease of measuring parameters: the resource requirement and impact of measuring parameters for each model, including cost, schedule impact for data collection, and physical significance of parameters to the software development process.
- Quality of assumptions: the closeness to the real world, and adaptability to a special environment.
- Capability: the ability of each model to estimate useful quantities needed by software project personnel, including expected MTTF, time to reach a specified MTTF goal, and the required resources to reach that goal.
- Applicability: the ability to handle program evolution and change in test and operational environment.

- Simplicity: ease of understanding the concept, data collection, program implementation, and validation.
- Insensitivity to noise: the ability of the model to produce results in spite of small differences in input data and parameters without losing responsiveness to significant differences

#### **6.1.1 Model Predictive Validity**

To compare a set of models on a given set of failure data, one must examine which of the **fitted** models is best in agreement with the observed data. **A fitted model** is one that has had its parameters estimated from the observed data. The question being asked is: Is it plausible to have obtained the observed data by sampling from the fitted model? If **F** is the function of the model with estimated parameters, this question can be answered by a hypothesis test with a null hypothesis:

 $H_0$ : the failure data are from a model with distribution function,  $\hat{\mathbf{F}}$ .

This is called a goodness-of-fit test since it tests how well the model "fits" the observed data. Goodness-of-fit tests are a way to detect systematically fairly gross disagreement between the data and the **fitted** model. The literature on goodness-of-fit tests is quite extensive; the chi-square and **Kolmogorov-Smirnov** tests are the most popular tests **[HOEL71].** 

In addition to these techniques for assessing model fit, the following four measures can be used to compare model forecasts on a set of failure data:

#### **6.1.1.1** Accuracy

Forecasting accuracy is measured by the prequential likelihood (PL) function [LITT86]. Let the observed failure data be a sequence of times  $t_1, t_2, ..., t_{i-1}$  between successive failures. The objective is to use the data to forecast the future unobserved  $T_i$ . More precisely, we want a good estimate of  $F_i(t)$ , defined as  $P(T_i, < t)$ , i.e., the probability that  $T_i$  is less than a variable t. The

forecasting distribution Pi(t) for  $T_i$  based on  $t_1, t_2, ..., t_{i-1}$  will be assumed to have a pdf (probability density function).

$$f_i(t) = \frac{d}{dt} \tilde{F}_i(t)$$

For such one-step-ahead forecasts of  $T_{j+1}$ ,... $T_{j+n}$ , the prequential likelihood is:

$$PL_n = \prod_{i=j+1}^{i+n} \tilde{f}_i(t_i)$$

Since this measure is usually very close to zero, its natural logarithm is frequently used for comparisons. Given two competing software reliability models A and B, the **prequential** likelihood ratio is given by

$$PLR_n = \frac{P \ln(A)}{P \ln(B)}$$

The ratio represents the likelihood that one model will give more accurate forecasts than the other model. If  $PLR_n \rightarrow \infty$  as  $n \rightarrow \infty$ , model A is favored over model B.

#### 6.1.1.2 Bias

A model is considered *biased* if it forecasts values that are consistently longer than the observed failure times, or consistently shorter than the observed times. To measure the amount of a model's bias, one can compute the maximum vertical distance (i.e., the Kolmogorov Distance [HOEL7 1]) between the line of unit slope and the values of the probability integral transformation given by:

$$u_i = \tilde{F}_i(t_i)$$

Each  $\mathbf{u_i}$  is a probability integral transform of the observed  $\mathbf{t_i}$  using the previously calculated predictor  $\mathbf{\tilde{F}_i}$  based upon  $\mathbf{t_1}, \mathbf{t_2}, \ldots, \mathbf{t_{i-1}}$ . That is,  $\mathbf{u_i}$  is the estimated model distribution function evaluated at the observed failure times. To identify the direction toward which a model is biased, use the notation that a positive number means that the model tends to be

optimistic, while a negative number represents the model to be pessimistic. In either case, the smaller the absolute value of the number is, the less bias there is inherent in the model.

#### 6.1.1.3 Trend

In some cases a model may be optimistic in an early set of forecasts and pessimistic in a later set of forecasts. The bias test described above will average these effects, and the model will appear unbiased. In this case, it is important to examine the **u**<sub>i</sub>'s for trend. Trend is defined as the Kolmogorov Distance between the line of unit slope and the cdf of **y**<sub>i</sub>, where

$$Xi = -\ln(1 - u_i)$$

$$y_{i} = \frac{\sum_{j=1}^{i} x_{j}}{\sum_{j=1}^{n} x_{j}} \text{ for } i \leq n$$

The trend represents the consistency of the model's bias. A small value means that the model is more adaptable to changes in the data behavior, and hence yields better performance.

#### 6.1.1.4 Noise

The test for noise is roughly analogous to the mean square error in classical statistics. The goal of the measure is to indicate objectively which model is giving the least variable forecasts (i.e., finding the most stable model for a particular data set), The measure is defined as

Noise = 
$$\sum_{i=2}^{n} \left| \frac{r_i - r_{i-1}}{r_{i-1}} \right|$$

where  $\mathbf{r_i}$  is the forecasted failure rate (1/ $\mathbf{T_i}$ ). Note that the forecasted median of the failure time distribution, denoted by  $\mathbf{m_i}$ , may be used in place of  $\mathbf{r_i}$ . In either case, small values represent less noise in the forecasts of the model, indicating better smoothness. A

Noise value equal to infinity indicates that a failure rate of zero has been forecasted by the model.

#### **6.1.2** Ease of Measuring Parameters

Ease of measuring parameters refers to the number of parameters a model requires, and the difficulties in estimating these parameters. Most software reliability estimation models incorporate either two or three parameters. As a rule-of-thumb, a model requires failure data equal to at least five times the number of parameteres to be estimated. In general, a three-parameter model can achieve better accuracy in fitting the failure data curve than can a two-parameter model. However, this is not generally true for making software reliability forecasts. When two models demonstrate the same level of forecasting capability, the model which requires fewer parameters should be chosen. This is not only because a model with fewer modes is easier to apply, but also because a software reliability engineer can more successfully interpret the physical significance of the parameters to provide appropriate feedback to the software development process.

#### **6.1.3 Quality of Assumptions**

The assumptions that a software reliability model makes should be as close to the real project testing and operational situation as possible. Common assumptions made in the software reliability models are:

- Test input randomly encounter faults.
- The effects of all failures are independent.
- The test space "covers" the use space.
- All failures are observed when they occur.
- Faults are immediately removed upon failure or not counted again.
- The software failure rate is related to the number of software faults remaining in the software; software reliability models specify this relationship.

If an assumption is testable, it should be **sup-**

ported by data to validate the assumption. If an assumption is not testable, it should be examined through the viewpoint of logical consistency and software engineering experience. Moreover, all model assumptions should be judged by their clarity and explicitness. This will help to determine whether a particular model applies to the current project.

#### **6.1.4** Capability

Capability refers to the ability of a model to estimate reliability related quantities for software systems. These quantities include:

- The present reliability of the software, the software failure rate, or mean-time-to-failure (MTTF), or the failure rate distribution.
- Confidence intervals for all estimated parameters.
- Expected date of achieving a specified reliability, failure rate, or MTTF objective.
- Resource (human and computer) and cost estimates related to achieving the reliability objective.

Other than the capability to make software reliability measurements in the testing and operational phase, the capability of a model to make software reliability predictions in the system design and early development phases is also very important. These predictions should be examined through future research in software metrics, the software development environment, and the operational profile.

#### 6.1.5 Applicability

Applicability of the software models should be examined through various sizes, structures, functions, and application domains. An advantage of a specific model is its usability in different development and operational environments, and different lifecycle phases. In the application of software reliability models, the following situations should be dealt with by the models:

Evolving software (i.e., software that is incrementally integrated during testing),

- Classification of failure severity,
- Incomplete failure data,
- Hardware execution rate differences,
- Multiple installations of the same software,
- Project environments departing from model assumptions.

#### 6.1.6 Simplicity

Simplicity refers to three aspects of a model: the data collection process, the modelling concept, and its implementation by a software tool. Simplicity in data collection reduces the measurement cost, increases the data accuracy, and makes it easier for model application. Simplicity in the modeling concepts makes it easier to understand the assumptions, estimate the parameters, apply the models, and interpret the results. Simplicity in the model implementation encourages an efficient use of computers to facilitate the model applications which are normally computationally intensive.

In choosing a model, one should give weight to simplicity. Until an organization has practiced reliability estimation a few times, no more complex models are warranted, nor in general will there be data to support more complex models.

#### 6.1.7 Insensitivity to Noise

Software reliability data generally contain noise irrelevant to the modeling process. The most common source of noise is that software failure data is recorded in project calendar time rather than in software execution time. Even when software failures are tracked carefully based on execution time, the software testing process may be inconsistent with the model assumptions (e.g., the software is not tested randomly). Therefore, a model should demonstrate its validity in an ideal situation as well as in situations when the failure data is incomplete or contains measurement uncertainties.

#### 6.2 Recommended Models

The following models are recommended as initial models for software reliability estimation; the order is arbitrary: the Schneidewind model, the generalized Exponential model, the Musa / Okumoto Logarithmic Poisson model and the Littlewood / Verrall model. If these models can not be validated (see Section 5.1.7) or do not meet the criteria defined in Section 6.1 for the project, alternative models are described in Appendix A.

### 6.2.1 Recommended Model: Schneidewind Model

#### **6.2.1.1** Schneidewind Objectives

The objectives of this model are to forecast the following software product attributes:

- Number of failures that will occur by a given time (execution time, labor time, or calendar time)
- Maximum number of failures that will occur over the life of the software
- Maximum number of failures that will occur after a given time
- Time required for a given number of failures to occur
- Number of faults corrected by a given time
- Time required to correct a given number of faults
- Number of outstanding (observed but not corrected) faults at a given time
- Incremental time required to correct a given number of outstanding faults
- Time required for outstanding faults to reach a given value

The basic philosophy of this model is that as testing proceeds with time, the failure detection process changes. Furthermore, recent failure counts are usually of more use than earlier counts in forecasting the future. Three approaches are employed in utilizing

the failure count data, i.e. number of failures detected per unit of time. Suppose there are m intervals of testing and  $\mathbf{f_i}$  failures were detected in the  $\mathbf{i^{th}}$  interval, one of the following can be done:

- Utilize all of the failures for the m intervals
- Ignore the failure counts completely from the first s-1 time intervals  $(2 \le s \le m)$  and only use the data from intervals s through m.
- Use the cumulative failure count from intervals 1 through s 1, i.e.

$$F_{s-l} = \sum_{i=1}^{s-l} f_i$$

The first approach is applicable when one feels that the failure counts from all of the intervals are useful in predicting future counts. The second approach is to be used when it is felt that a significant change in the failure detection process has occurred and thus only the last m - s + 1 intervals are useful in future failure forecasts. The last approach is an intermediate one between the other two. Here it is felt that the combined failure counts from the first s - 1 intervals and the individual counts from the remaining are representative of the failure and detection behavior for future forecasts.

#### **6.2.1.2** Schneidewind Assumptions

The assumptions to the Schneidewind model are:

- The number of failures detected in one interval is independent of the failure count in another.
- Only new failures are counted.
- The fault correction rate is proportional to the number of faults to be corrected.
- The software is operated in a similar manner as the anticipated operational usage.
- The mean number of detected failures

decreases from one interval to the next.

- The intervals are all the same length.
- The rate of failure detection is proportional to the number of faults within the Program at the time of test. The failure detection process is assumed to be a nonhomogeneous Poisson process with an exponentially decreasing failure detection rate. The rate is taken to be of the form

$$d_i = \alpha \exp(-\beta i)$$

for the  $i^{th}$  interval where a > 0 and b > 0 are the constants of the model.

#### **6.2.1.3** Schneidewind Structure

Two parameters are used in the model:  $\alpha$ , which is the failure rate at time m=0, and  $\beta$ , which is a proportionality constant that affects the failure rate over time (i.e., small  $\beta$  implies a large failure rate; large  $\beta$  implies a small failure rate). In **these** estimates: m is the last observed count interval; s is an index of time intervals;  $X_k$  is the number of observed failures in interval k,  $X_{s-1}$  is the number of failures observed from 1 through s-l intervals;  $X_{s,m}$  is the number of observed failures from interval s through m; and  $X_m = X_{s-1} + X_{s,t}$ . The likelihood function is then developed as

$$\begin{split} \log L &= X_m \Big[ \log X_m - 1 - \log(1 - \exp(-\beta m)) \Big] \\ &+ X_{s-1} \Big[ \log \Big( 1 - \exp(-\beta (s-1)) \Big) \Big] \\ &+ X_{s,m} \Big[ \log \Big( 1 - \exp(-\beta) \Big) \Big] \\ &- \beta \sum_{k=0}^{m-s} (s+k-1) X_{s+k} \end{split}$$

This function is used to derive the equations for estimating a and  $\beta$  for each of the three approaches described earlier. In the equations that follow, a and  $\beta$  are estimates of the population parameters.

#### **Parameter Estimation: Approach 1**

Use all of the failure counts from interval 1 through m (i.e., s=1). The following two equations are used to estimate  $\beta$  and a, respectively.

$$\frac{1}{\exp(\beta)-1} - \frac{m}{\exp(\beta m)-1} = \sum_{k=0}^{m-1} k \frac{X_{k+1}}{X_m}$$

$$\alpha = \frac{\beta X_{\rm m}}{1 - \exp(-\beta m)}$$

#### **Parameter Estimation: Approach 2**

Use failure counts only in intervals s through m (i.e.,  $1 \le s \le m$ ). The following two equations are used to estimate  $\beta$  and  $\alpha$ , respectively. (Note that approach 2 is equivalent to approach 1 for s = 1.)

$$\frac{1}{\exp(\beta)-1} \frac{m-s+1}{\exp(\beta(m-s+1))-1} = \sum_{k=0}^{m-s} k \frac{X_{k+s}}{X_{s,m}}$$

$$\alpha = \frac{\beta X_{s,m}}{1 - \exp(-\beta(m-s+1))}$$

#### **Parameter Estimation: Approach 3**

Use cumulative failure counts in intervals 1 through s-l and individual failure counts in intervals s through m (i.e.,  $2 \le s \le m$ ). This approach is intermediate to approach 1 which uses all of the data and approach 2 which discards "old" data. The following two equations are used to estimate  $\beta$  and a, respectively. (Note that approach 3 is equivalent to approach 1 for s = 2.)

$$\frac{(s-1)X_{s-1}}{\exp(\beta(s-1))-1+\exp(\beta)-1} = \frac{mX_{m}}{\exp(\beta m)-1}$$

$$= \sum_{k=0}^{t-s} (s+k-1)X_{s+k}$$

$$\alpha = \frac{\beta X_m}{1 - \exp(-\beta m)}$$

#### Mean Square Error Criterion

The Mean Square Error (MSE) criterion can be used to select one of the three approaches by finding the optimal value of s. The MSE computes the sum of the squared differences between model predictions and actual cumulative failure counts x(i) in the range  $s \le i \le m$ . The following equation applies to approach 2 above. For approach 1 and approach 3, s=1.

$$MSE = \frac{\sum_{i=s}^{m} \left[ a / \beta \left( 1 - \exp(-\beta (i - s + 1)) \right) - x(i) \right]^{2}}{m - s + 1}$$

Thus, for each value of s, compute MSE using the above formula. Choose s equal to the value for which MSE is smallest. The result is an optimal triple  $(\beta, a, s)$  for your data set. Then apply the appropriate approach to your data.

#### 6.2.1.4 Schneidewind Limitations

The limitations of the model are the following:

- It does not account for the possibility that failures in different intervals may be related.
- It does not account for repetition of failures.
- It uses intervals of equal length.
- It does not account for the possibility that failures can increase over time as the result of software modifications.

These limitations can be ameliorated by configuring the software into versions that represent the previous version plus modifications. Each version represents a different module for reliability prediction purposes: the model is used to predict reliability for each module.

### **6.2.1.5** Schneidewind Data Requirements

The only data requirements are the number of errors,  $f_{i,}$   $i = 1, \ldots, m$ , per testing period.

Although a data base is not required, it would be very useful to create and maintain a reliability data base for several reasons: input data sets could be rerun, if necessary; reliability predictions and assessments could be made for various projects; predicted reliability could be compared with actual reliability for these projects. This data base would allow the model user to perform several useful analyses: to see how well the model is performing; to compare reliability across projects to see whether there are development factors that contribute to reliability; and to see whether reliability is improving over time for a given project or across projects.

#### 6.2.1.6 Schneidewind Applications

The major model applications are described below. These are separate but related uses of the model that, in total, comprise an integrated reliability program.

- Forecasting: Forecasting future failures, fault corrections and related quantities described in section 6.2.1.7.
- Control: Comparing forecast results with pre-defined goals and flagging software that fails to meet those goals.
- Assessment: Determining what action to take for software that fails to meet goals (e.g., intensify inspection, intensify testing, redesign software, revise process). The formulation of test strategies is also part of assessment. Test strategy formulation involves the determination of: priority, duration and completion date of testing, allocation of personnel, and allocation of computer resources to testing.

#### **6.2.1.7** Reliability Forecasts

Using the optimal triple  $(\alpha, \beta, s)$  which were given in section 6.2.1.3, various reliability

forecasts can be computed. The approach 2 equations are given where  $T \ge s$ . For approach 1 and approach 3, s=1 and  $T \ge 1$ , where T is preferably execution time but can be labor time or calendar time.

• Time to detect a total of F failures, when the current time is t and X(t) failures have been observed

$$T_f(t) = log[\alpha / (a - \beta(F(t) - X(t))) / \beta] - (t-s + 1)$$
  
for  $a > \beta(F(t) + X(t))$ 

• Forecasted Number of Failures after time T

$$F(T) = (a / \beta)[1 - \exp(-\beta(T - s + 1))]$$

• Maximum Number of Failures  $(T = \infty)$ 

$$F(\infty) = \alpha / \beta$$

 Maximum Number of Remaining Failures, forecasted at time t, after X(t) failures have been observed

$$RF(t) = a / P-X(t)$$

• Faults Corrected after time T

$$C(T) = (\alpha / \beta) [1 - \exp(-\beta((T - s + 1) - At))]$$

where At is the mean lag in correcting faults after failures have been observed. (At can be estimated from the data.)

• Time to correct C faults

$$T_{C} = \Delta t + \left[ \left( \log \left[ \alpha / (\alpha - \beta C) \right] \right) / \beta \right] + s - 1$$
  
for  $\alpha > \beta C$ 

· Outstanding Faults Remaining at time T

$$N(T) = F(T) \cdot C(T)$$

· Outstanding Fault Correction Time

$$\Delta T_{N} = \left[ \log \left( \left( N\beta \exp \left( \beta (T - s + 1) \right) / \alpha \right) + 1 \right) \right] / \beta$$

where N is the number of faults to correct starting at time T.

Outstanding Fault Time

The predicted time for the number of outstanding faults to reach the value N is

$$T_{N} = \left[ \left( \log \left[ \left( \alpha \exp \left( \beta \Delta T_{N} \right) - 1 \right) / \beta N \right] \right) / \beta \right] + s - 1$$

# **6.2.1.8** Schneidewind Implementation Status and Reference Applications

The model has been implemented in FORTRAN by the Naval Surface Warfare Center, Dahlgren, Virginia as part of the Statistical Modeling and Estimation of Reliability Functions for Software (SMERFS). It can be run on an IBM PC (or compatible) or DEC VAX and is available on DOS diskette or magnetic tape, respectively.

Known applications of this model are:

- IBM, Houston, Texas: Reliability prediction and assessment of the on-board NASA Space Shuttle software [SCHN92]
- Naval Surface Warfare Center, Dahlgren, Virginia: Research in reliability prediction and analysis of the TRIDENT I and II Fire Control Software [FARR91]
- NASA JPL, Pasadena, California: Experiments with multi-model software reliability approach [LYU92]
- Hughes Aircraft Co., Fullerton, California: Integrated, multi-model approach to reliability prediction [BOWE87]

### **6.2.2 Recommended Model:** Generalized Exponential Model

### **6.2.2.1 Generalized Exponential Objectives**

Many popular software reliability models yield simular results. The basic idea behind the generalized exponential model is to simplify the modeling process by using a single set of equations to represent models having exponential hazard functions.

The generalized exponential model contains the ideas of several well-known software reliability models. The main idea is that the failure **occurence** rate is proportional to the number of faults remaining in the software. Furthermore, the failure rate remains constant between failure detections and the rate is reduced by the same amount after each fault is removed from the software. Thus, the correction of each fault has the same effect in reducing the hazard of the software. The objective of this model is to generalize the forms of several well-known models into a form that can be used to forecast:

- Number of failures that will occur by a given time (execution time, labor time, or calendar time)
- Maximum number of failures that will occur over the life of the software
- Maximum number of failures that will occur after a given time
- Time required for a given number of failures to occur
- Number of faults corrected by a given time
- Time required to correct a given number of faults

### **6.2.2.2** Generalized Exponential Assumptions

The basic assumptions of the Generalized Exponential Model are:

- The failure rate is proportional to the current fault content of a program.
- All failures are equally likely to occur and are independent of each other.
- Each failure is of the same order of severity as any other failure.
- The software is operated in a similar manner as the anticipated operational usage.

• The faults which caused the failure arc corrected instantaneously without introduction of new faults into the program.

### **6.2.2.3** Generalized Exponential Structure

The Generalized Exponential Structure begins with a simple, but relatively general, form for the software hazard function:

$$z(x) = K[E_o - E_c(x)]$$

where

- **x** = a time or resource variable which gauges the progress of the project.
- E<sub>0</sub> = the initial number of faults in the program which will lead to failures. It can also be viewed as the number of failures which would be experienced if testing continued indefinitely.

- **E**<sub>c</sub> = the number of faults in the program which have been found and corrected once x units of time or effort have been expended
- K = a constant of proportionality; failures per resource or time units, per fault remaining

Inspection of this equation shows that the number of remaining faults,  $\mathbf{E_r}$ , is given by

$$E_r = z(x) / K = [E_o - E_c(x)]$$

Note that this equation has no fault generation term; it assumes that no new faults which will lead to failures are generated during program debugging. More advanced models that include fault generation are discussed in [MUSA87] and [SHOO83].

Many models in common use can be represented by the above set of equations with various assumptions regarding the

Table 1 Common Reliability Models that Fit the Generalized Exponential Form for the Failure Rate Function

MODELNAME	ORIGINAL HAZARD FUNCTION	PARAMETER EQUIVALENCES	COMMENTS
Generalized Form	$K[E_0 - E_c(x)]$	•	
Exponential model [SHOO72]	$K'[E_0/I_T - \varepsilon_c(x)]$	$\varepsilon_{c} = E_{c} / I_{T}$ $K = K' / I_{T}$	Normalized with respect to I <sub>T</sub> , the number of instructions
Jelinski-Moranda [JELI72]	φ[N-(i-1)	$ \phi = K  N = E_0  E_c = (i-1) $	Applied at the discovery of an error and before it is corrected
Basic Model [MUSA76]	$\lambda_0[1 - \mu/\nu_0]$	$\lambda_{o} = KE_{o}$ $\nu_{o} = E_{o}$ $\mu = E_{c}$	If the same assumptions are used to predict m and $\mathbf{E_c}$ , then this <b>model</b> and the exponential model are the same.
Logarithmic [MUSA83]	λ <sub>o</sub> exp (-φμ)	$\lambda_{o} = KE_{o}$ $E_{o} \cdot E_{c}(x)$ $= E_{o} \exp(-\phi \mu)$	Basic assumption is that the remaining number of errors decreases exponentially.

parameters and the form that the fault correction function,  $\mathbf{E_0(x)}$ , takes. Some of these models are summarized and compared in Table 1. In the original development of each model in this table, one or more time or resource variables were used. In retrospect, all of the models can, in general, be phrased in terms of any of the time or resource variables given in Table 1. Thus, unless stated to the contrary, the use of a specific time or resource variable does not differentiate one model from another.

Given the data defined in section 6.2.2.5, estimation of any of the model parameters given in Table 1 reduces to a problem in statistical parameter estimation [HOEL7 1] or [SHOO90a]. There are three basic methods: moments, least squares, and maximum likelihood. Although the original developments of the various models or some of the computer tools available to support these models may have used only one or two of these methods, all three are applicable to each of the models.

$$z(x) = \frac{\text{Failures } (x) - \mathbf{n-r}}{\text{Hours } (x) - \mathbf{H}}$$
 (6.1)

where

$$H = \sum_{i=1}^{n-r} t_i + \sum_{i=1}^{r} T_i$$

Equating the unified form equation with equation (6.1) at two different values of time

yields

$$z(x_1) = \frac{n_1 - r_1}{H_1} = K[E_o - E_c(x_1)]$$
 (6.2)

$$z(x_2) = \frac{n_2 - r_2}{H_2} = K[E_o - E_c(x_2)]$$
 (6.3)

Simultanaeous solution of these two sets of equations, equations (6.2) and **(6.3)**, yields estimators denoted by ^, for the parameters.

$$\hat{\mathbf{E}}_{o} = \frac{\mathbf{E}_{c-1} - \frac{\mathbf{z}(\mathbf{x}_{1})}{\mathbf{z}(\mathbf{x}_{2})} \mathbf{E}_{c}(\mathbf{x}_{2})}{1 - \frac{\mathbf{z}(\mathbf{x}_{1})}{\mathbf{z}(\mathbf{x}_{2})}}$$

$$= \frac{\mathbf{z}(\mathbf{x}_{2} \ \mathbf{E}_{c}(\mathbf{x}_{1} - \mathbf{z}(\mathbf{x}_{1} \ \mathbf{E}_{c}(\mathbf{x}_{2}) - \mathbf{z}(\mathbf{x}_{1}))}{\mathbf{z}(\mathbf{x}_{2}) - \mathbf{z}(\mathbf{x}_{1})}$$
(6.4)

$$\hat{K} = \frac{z(x_1)}{\hat{F}_o - E_c(x_1)}$$

$$= \frac{z(x_2) - z(x_1)}{E_c(x_1) - E_c(x_2)}$$
 (6.5)

Since all of the parameters of the five models in Table 1 are related to Eo and K by simple transformations, equations (6.4) and (6.5) along with the transformations (parameter equivalences) hold Thus these equations can be used to obtain moment estimates for all the models. For example, we could start using the Musa Basic model of Table 1 and apply the moment estimate procedure to determine  $\hat{\lambda}_{o}$  and  $\hat{v}_{o}$  in an analogous fashion to what was done in equations (6.2) and (6.3). More simply, we could use equations (6.4) and (6.5) and the transformation  $v_{o} = E_{o}$  and  $\mu_{o} = KE_{o}$  to obtain

$$\hat{v}_{o} = \hat{E}_{o} - \frac{z(x_{2})E_{c}(x_{1}) - z(x_{1} E_{c}(x_{2}))}{z(x_{2}) - z(x_{1})}$$

$$\hat{\lambda}_{o} = \hat{K}\hat{E}_{o}$$

$$\frac{z(x_2) - z(x_1)}{E_c(x_1) - E_c(x_2)} \times \frac{z(x_2)E_c(x_1) - z(x_1)E_c(x_2)}{z(x_2) - z(x_1)}$$

$$\hat{\lambda}_{o} = \frac{z(x_{2})E_{c}(x_{1}) - z(x_{1})E_{c}(x_{2})}{E_{c}(x_{1}) - E_{c}(x_{2})}$$

Which are the moment estimation equations. Similar results can be obtained for the other models in Table 1. More advanced estimates of the model parameters can be developed using least squares and maximum likelihood estimation theory ([SHOO90a].

### **6.2.2.4** Generalized Exponential Limitations

The generalized exponential model has the following limitations:

- It does not account for the possibility that each failure may be dependent on others
- It assumes no new faults are introduced in the fault correction process
- Each fault detection may have a different impact on the software when the fault is corrected. The Logarithmic model handles this by saying that earlier fault corrections have a greater impact than later ones.
- It does not account for the possibility that failures can increase over time as the result of program evolution, although techniques for handling this limitation have been developed.

### 6.2.2.5 Generalized Exponential Data Requirements

During test, a record will be made of each of the total of **n** test runs. The test results include the r successes and the n-r failures along with the time of occurrence measured in terms of clock time and operational execution time, or test time if operational tests are unavailable. Additionally, there should be a record of the times for the r successful

runs. Thus, the desired data is the total number of runs n, the number of successful runs r, the sequence of clock times to failure t1, t2, ..., t<sub>n-r</sub> and the sequence of clock times for runs without failure T1, T2, .... T<sub>r</sub>. All the times should be for actual or simulated operation; however, if only test time is available, that should be recorded. A description is needed along with the data describing whether is represents operation, simulated operation, or test and the circumstances and conditions governing the input data. If possible, a similar set of operational data should be recorded.

### **6.2.2.6** Generalized Exponential Applications

The Generalized Exponential Model(s) tend to be optimistic. It is applicable when the operational profile is "regular," and the software debugging process is well controlled (i.e., the fault correction process tends to be complete and not error prone.)

The major model applications are described below. These are separate but related uses of the model that, in total, comprise an integrated reliability program.

- Forecasting: forecasting future failures, fault corrections, and related quantities described in section 6.2.2.7.
- Control: comparing forecast results with predefined goals and flagging software that fails to meet those goals.
- Assessment: determining what action to take for software that fails to meet goals (e.g., intensify inspection, intensify testing, redesign software, revise process). The formulation of test strategies is also part of the assessment. Test strategy formulation involves the determination of: priority, duration, and completion date of testing, allocation of personnel, and allocation of computer resources to testing.

#### **6.2.2.7** Reliability Forecasts

Besides the estimate of the total number of faults given by  $\hat{\mathbf{E}}_{\mathbf{0}}$ , other estimates are:

• Estimated time to remove the next m faults

$$= \sum_{j=n+1}^{n+m} \frac{1}{\hat{K}_0(\hat{E}_0 - j + 1)}$$

• Estimate of the current failure rate at time

$$\tau = \hat{\mathbf{K}}_{o} \left( \hat{\mathbf{E}}_{o} \exp(-\hat{\mathbf{K}}_{o} \tau) \right)$$

For other quantities that can be estimated, see the references listed in paragraph 6.2.2.8.

# **6.2.2.8** Generalized Exponential Implementation Status and Reference Applications

The Generalized Exponential Model has not been implemented as a standalone model. The many models it represents, however, have been implemented in several tools including SMERFS from the Naval Surface Warfare Center, Dahlgren, VA, Software Reliability Modeling Program (SRMP) from the the Center for Software Reliability in London, England, and RELTOOLS from AT&T. See Appendix B for details.

While the generalized exponential model has not been used widely, many of the specific models that it covers as special cases have been applied successfully. See the following for example applications:

- Jelinski Z.and Moranda, P. B., "Software Reliability Research," W. Freiberger, Editor, Statistical Computer Performance Evaluation, Academic Press, New York, pp. 465-484.
- Shooman, M. L. and Richeson, G., "Reliability of Shuttle Control Center Software," Proceedings Annal Reliability and Maintainabilty Symposium, January 1983, pp. 125-135.
- Kruger, G. A., "Validation and Further Application of Software Reliability Growth Models," Hewlett-Packard Journal, April 1989, pp. 75-79.

# 6.2.3 Recommended Model: Musa / Okumoto Logarithmic Poisson Execution Time Model

#### 6.2.3.1 Musa / Okumoto Objectives

The logarithmic Poisson is especially applicable when the testing is done according to an operational profile that is very nonuniform in nature. Early fault corrections have a larger impact on the failure intensity function than later ones. The failure intensity function tends to be convex with decreasing slope for this situation. Thus a logarithmic Poisson model may be very appropriate for this circumstance.

If one is also interested in relating calendar time considerations (e.g., completion of testing, resource management, etc.) to reliability, the logarithmic Poisson is the only non-exponential model that can do this at this time.

Considerations relating to computer utilization, personnel level, and current and projected failure rate trade-offs can be performed to balance reliability considerations with time and resource constraints.

The number of failures occurring over an infinite amount of time is unbounded for this model [MUSA87]. It is especially applicable when high nonuniformity is experienced in the operational profile. The belief is that as one detects the earlier faults a greater reduction in the failure intensity is experienced. With a highly non-uniform profile exhibited, early fault corrections make a more substantial impact on the failure behavior of the software than later ones. This behavior of the failure intensity can be more adequately modeled by a logarithmic Poisson approach.

If there is a decreasing effectiveness of the repair process, then this model can yield an unbounded number of failures even though the number of faults may be finite.

### **6.2.3.2** Musa / Okumoto Assumptions

The specific assumptions for this model are:

• The software is operated in a similar

manner as the anticipated operational usage.

- · Failures are independent of each other.
- The failure intensity decreases exponentially with the expected failures experienced.

Note: There are two consequences of the third assumption. First, the expected number of failures is a logarithmic function of time. Second, the model may report an infinite number of failures.

#### 6.2.3.3 Musa / Okumoto Structure

From the model assumptions we have:

h(z) = failure rate function after t amount of execution time has been expended

$$= \lambda_0 \exp[-\theta \mu(\tau)]$$

The parameter  $\lambda_0$  is the initial failure rate function and  $\theta$  is the failure rate decay **pa**-rameterwith  $\theta > 0$ .

Using a reparameterization of  $\beta_o = \theta^{-1}$  and  $\beta_1 = \lambda_o \theta$ , then the maximum likelihood estimates of  $\beta_o$  and  $\beta_1$  are shown in [MUSA87] to be the solutions of the following equations:

$$\theta = \hat{\beta}_{o} - \frac{n}{\ln(1 + \hat{\beta}_{1}t_{n})}$$

$$\theta = \frac{1}{\hat{\beta}_1} \sum_{i=1}^{n} \frac{1}{1 + \hat{\beta}_1 t_n} - \frac{nt_n}{\left(1 + \hat{\beta}_1 t_n\right) \ln\left(1 + \hat{\beta}_1 t_n\right)}$$

Here  $t_n$  is the cumulative CPU time from start to the current time. Over this period, we have observed a total of n failures. Once maximum likelihood estimates are found for  $\beta_o$  and  $\beta_1$ , the maximum likelihood estimates for  $\theta$  and  $\lambda_o$  are, using the invariance property of such estimators:

$$\hat{\theta} = \frac{1}{n} \ln (1 + \hat{\beta}_1 t_n)$$
 and

$$\hat{\lambda}_{o} = \hat{\beta}_{o} \hat{\beta}_{1}$$

#### 6.2.3.4 Musa / Okumoto Limitations

Two limitations are:

- The failures may not be independent of one another.
- The failure intensity may rise as modifications are made to the software.

### 6.2.3.5 Musa / Okumoto Data Requirements

The required data is either:

- . The time between failures, i.e., the X i's.
- The time of the failure occurrences, i.e.,

$$\mathbf{t_i} = \sum_{j=1}^{l} X_j$$

### **6.2.3.6** Musa / Okumoto Applications

The major model applications are described below. These are separate but related applications that, in total, comprise an integrated reliability program.

- Prediction: Estimating future failure times, fault corrections, and related quantities described in Musa's book [MUSA87].
- Control: Comparing prediction results with pre-defined goals and flagging software that fails to meet goals.
- Assessment: Determining what action to take for software that fails to meet goals (e.g., intensify inspection, intensify testing, redesign software, revise process). The formulation of test strategies is also a part of assessment. It involves the determination of: priority, duration and completion date of testing, and allocation of personnel and computer resources to testing.

# 6.2.3.7 Reliability Forecasts

In their book, Musa, Iannino, and Okumoto [MUSA87] show that from the assumptions above and the fact that the derivative of the mean value function is the failure rate function, we have:

$$\hat{\lambda}(\tau) = \frac{\hat{\lambda}_o}{\hat{\lambda}_o \theta \tau + 1}$$

 $\hat{\mu}(\tau)$  = mean number of failures experienced by the time  $\tau$  is expended

$$= \frac{1}{\hat{\boldsymbol{\theta}}} \ln (\hat{\boldsymbol{\lambda}}_{\boldsymbol{o}} \hat{\boldsymbol{\theta}} \boldsymbol{\tau} + 1)$$

The estimates of additional reliability measures are provided in the references listed in paragraph 6.2.3.8.

# 6.2.3.8 Musa / Okumoto Implementation Status and Reference Applications

The model has been implemented by the Naval **Surface** Warfare Center, Dahlgren, VA as part of SMERFS. It can be run on any computer system with a FORTRAN compiler and is available upon request.

This model has also been implemented in the set of programs written by AT&T (see Appendix B for details).

This model has been applied widely. See the following for example applications:

- . Musa, J. D., Iannino, A., and Okumoto, K., Software Reliability: Measurement, Prediction, and Application, New York, McGraw-Hill, 1987.
- Musa, J. D. and Okumoto, K., "A Logarithmic Poisson Execution Time Model for Software Reliability Measurement," Proceedings of the 7th International Conference on Software Engineering, Orlando, FL, 1984, pp. 230-238.
- . Ehrlich, W. K., Stampfel, J. P., and Wu, J. R., "Application of Software Reliability

Modeling to Product Quality and Test Process," Proceedings of the **IEEE/TCSE** Subcommittee on Software Reliability Engineering Kickoff Meeting, NASA Headquarters, Washington, DC, April 1990, paper 13.

# **6.2.4 Recommended Model:** Littlewood / Verrall Model

# **6.2.4.1** Littlewood / Verrall Objectives

The intention of the Littlewood / Verrall is to model the doubly stochastic nature of the software failure process. There are two basic sources of uncertainty which need to be taken into account when software fails and fixes are attempted.

In the first place there is uncertainty about the nature of the operational environment: we do not know when a certain input will show itself, and in particular we do not know which inputs will be selected next. Thus, even if we had complete knowledge of which inputs were failure-prone (and of course this is never the case), we still could not tell with certainty when the next one to induce a failure would be received. All software reliability models recognize this source of uncertainty, and it is often presented mathematically by a simple Poisson process: i.e., it is assumed that failures occur **purely randomly**. This means the time to next failure, for example, will have an exponential distribution.

The second source of uncertainty concerns what happens when an attempt is made to remove the fault that caused the failure. The aforementioned models that assume that the process of failures is locally purely random, it is this uncertainty that governs the changes in the failure rate as debugging proceeds: i.e., it determines the nature of the **reliability growth.** There is uncertainty here for two main reasons. In the first place, it is clear that not all the faults contribute the same amount to the unreliability of a program. Some contribute a greater amount than others. If the software has failed because a fault has been detected that contributes a large amount to the overall reliability, then there will be a correspondingly large increase in the

reliability (reduction in the failure rate) when this is removed. In the second place, we can never be sure that we actually have removed a fault successfully; indeed it is possible that some new fault has been introduced and the reliability of the program made worse. The result of these two effects is that the failure rate of a program changes in a random way as debugging proceeds: there will likely be a downwards jump in this rate at each fix attempt, but this is not certain, and the size of the jump in unpredictable.

The Littlewood / Verrall model, unlike the other models discussed, takes account of both of these sources of uncertainty in the failure process • that due to basic unpredictability of the environment which profers inputs for execution, and that due to an intrinsic uncertainty of the effects of the human activities during debugging.

# **6.2.4.2** Littlewood / Verrall Assumptions

The following assumptions apply to the Littlewood / Verrall model:

- The software is operated during the collection of failure data in a manner that is similar to that for which predictions are to be made; the test environment is an accurate representation of the operational environment.
- The times between successive failures are conditionally independent exponential random variables, i.e., locally (between failures) the failure process is purely random.
- The fixing process involves uncertainty represented by allowing the successive failure rates, following successive fix attempts, to be a sequence of independent random variables.

# **6.2.4.3** Littlewood / Verrall Structure

This model treats the successive rates of occurrence of failures as fixes take place, as random variables. It assumes

$$P(t_i | \Lambda_i = \lambda_i) = \lambda_i e^{-\lambda_i t_i}$$

The sequence of rates  $\lambda_i$  is treated as a sequence of independent stochastically decreasing random variables. This reflects the likelihood, but not certainty, that a **fix** will be effective. It is assumed that

$$g(\lambda_i) = \frac{\psi(i)^{"} \lambda_i^{\alpha-1} e^{-\psi(i)\lambda_i}}{\Gamma(\alpha)}$$
 for  $\lambda_i > 0$ 

which is a gamma distribution with parameters  $\alpha, \psi(i)$ .

The function  $\psi(i)$  determines the reliability growth. If, as is usually the case,  $\psi(i)$  is an increasing function of i, it is easy to show that  $\Lambda_i$  forms a stochastically decreasing sequence. For this model a fix may make the program less reliable, and even if an improvement takes place it is of uncertain magnitude.

By setting  $\psi(i)$  to either  $\beta_o + \beta_1 i$  or  $\beta_o + \beta_1 i^2$  and eliminating a, Littlewood and Verrall present a method of estimating  $\beta_o$  and  $\beta_1$  based upon maximum likelihood. By eliminating a from the likelihood equations; i.e., the estimate of a can be expressed as a function of the estimates of the other two parameters. See [FARR83,LITT73] for details. The maximum likelihood calculation needs to be done using a numerical optimization routine which is available in commercially available software, such as those found in Appendix B.

Least squares estimates of the parameters  $(\alpha, \beta_0, \beta_1)$  are found by minimizing:

$$S(\alpha, \beta_0, \beta_1) = \sum_{i=1}^{n} \left( x_i - \frac{\left[ \psi(i) \right]}{\alpha - 1} \right)^2$$

See [FARR83] for further details.

# 6.2.4.4 Littlewood / Verrall Limitations

The primary limitation as with all Bayesian analysis is the specification of the prior

density function  $g(\lambda_i)$ . A secondary limitation of **the** Littlewood / Verrall model is **that** it cannot estimate the number of faults remaining in the software (the estimate may be infinite depending on the  $\psi(i)$  function).

# 6.2.4.5 Littlewood / Verrall Data Requirements

The only required data is either:

- The time between failures, i.e. the  $X_i$ 's.
- The time of the failure occurrences, i.e.

$$t_i = \sum_{i=1}^i X_i$$

# **6.2.4.6** Littlewood / Verrall Applications

The Littlewood / Verrall Model (or Inverse Polynomial Model) is a conservative and pessimistic model. It is applicable when the operational profile is non-uniform and even irregular, especially when the software debuging process is imperfect (i.e., the fault correction process tends to be incomplete or error-prone), This model has the capability of adjusting the parameters to reflect the situation.

The major model applications are described below. These are separate but related uses of the model that, in total, comprise an integrated reliability program.

- Forecasting: Forecasting future failures, fault corrections, and related quantities described in section 6.2.1.7.
- Control: Comparing forecast results with pre-defined goals and flagging software that fails to meet those goals.
- Assessment: Determining what action to take for software that fails to meet goals (e.g., intensify inspection, intensify testing, redesign software, revise process).
   The formulation of test strategies is also part of assessment. Test strategy formulation involved the determination of: priority,

duration and completion date of testing, allocation of personnel, and allocation of computer resources to testing.

### 6.2.4.7 Reliability Forecasts

Estimation of reliability and other associated terms is via substitution of the parameter estimates into appropriate expressions. An estimate of the Mean Time To Failure, (MTTF), is:

$$MTTF = \hat{E}(X_i) = \frac{[\hat{\psi}(i)]}{\hat{\alpha} - 1}$$

The expression for failure rate is:

$$\hat{\lambda}(t) = \frac{\hat{\alpha}}{\left(t + \hat{\psi}(i)\right)}$$

(Note that the failure rate expression is a continuously decreasing function during periods of failure-free working, representing the greater confidence that comes from such evidence)

The reliability function is:

$$R(t) = P(T_i > t) = \hat{\psi}(i)^{\hat{\alpha}}[t + \hat{\psi}(i)]^{-\hat{\alpha}}$$

In all of the above expressions,  $\hat{\psi}(i)$  and  $\hat{\alpha}$  are the estimates of the two respective parameters from section 6.2.4.3.

For other quantities that can be estimated, see the references listed in paragraph 6.2.4.8.

# 6.2.4.8 Littlewood / Verrall Implementation Status and Reference Applications

The model has been implemented as part of the SMERFS. It can be run on any computer system with a FORTRAN compiler and is available upon request.

This model has also been implemented in the Software Reliability Modeling Programs (SRMP) at the Center for Software Reliability in London, England by Dr. Littlewood and his associates of Reliability and Statistical

Consultants, Ltd. This program package runs in a PC environment.

The **Littlewood/Verrall** model has been applied widely. See the following for examples of applications:

- K. Kanoun, J. Sabourin, (1987), "Software Dependability of a Telephone Switching System," Proceedings 17th IEEE Symposium on Fault-Tolerant Computing (FTCS-17), Pittsburgh, PA.
- Mellor, P., (1986), "State of the Art Report on Software Reliability," *Infotech*, London
- Abdel-Ghaly, A. A., Chan, P. Y. and Littlewood, B., (1986), "Evaluation of Competing Software Reliability Predictions," IEEE Transactions of Software Engineering, SE-12 (9), 950-967

# **6.3 Experimental Approaches**

Several improvements to the software reliability models described in the previous sections have been recently proposed. First, researchers at the City University of London have devised a method of recalibrating the models [BROC92] to reduce their biases (see section 6.1.1.2). These findings to date suggest that the recalibrated models yield consistently more accurate forecasts than the uncalibrated models. Second, work has also been done in combining the results from two or more models in a linear fashion to increase predictive accuracy [LYU92, LU92]. This work suggests that such combinations yield more accurate results than individual models. The advantage of combining model results is the simplicity with which the combinations are formed – the models in the combination are executed individually, with only the results being combined. Third, efforts to incorporate software complexity metrics into reliability models [KAFU87, KHOS91], and to gauge the effects of different types of testing (e.g., branch testing, data path testing) on reliability growth [MATH92] are being investigated. Finally, the use of neural networks for software reliability parameter estimation is being investigated [KARU92].

Although these efforts show promise in

increasing the forecasting accuracy of software reliability modeling, there is not sufficient evidence to classify them as recommended practice at this time. They are included here to indicate some of the current avenues of investigation. Further experience with these methods may lead to their being classified as recommended practice in the future.

# 7.0 SOFTWARE RELIABILITY DATA

A variety of applications for software reliability measurement were described in Section 5 of this document. Section 6 provided a list of selection criteria as well as a set of models for estimating the reliability of the software product. Data collection provides the foundation on which both of these sections depend. This section addresses (1) a procedure for collecting data, (2) two data types, (3) the relationships between the two types, and (4) the **AIAA** data base **hierachy**.

### 7.1 Data Collection Procedure

The following nine steps can be used to establish a software reliability data collection process:

• Step 1: Establish the objectives.

The first step in planning to collect data is to determine the objectives of the data and what data items will be collected. Data collection does involve cost, so each item should be examined to see if the need is worth the cost. This should be done in the context of the planned application or applications of software reliability engineering. **If** the item is questionable, consider alternatives such as approximating the item or collecting it at a lower frequency. Look for possibilities of collecting data items that can serve multiple purposes. If this careful examination is not performed, the unnecessary burden in effort and cost on the project can result in the degradation of all data or even the abandonment of the effort.

• Step 2: Plan the data collection process.

It is recommended that all parties (designers, coders, testers, users, and key management) participate in the planning effort. The data collectors must be motivated if quality data is to be collected. Present the goals of the data collection effort. Relate it to their direct personal benefit. This will insure that all parties understand what is being done and the impact it will have on their respective organizations.

It is suggested that a first draft data collection plan be presented as a starting point. The plan should include topics such as:

- What data items will be gathered?
- Who will gather the data?
- How often will the data be reported?
- Formats for data reporting (e.g., electronic spreadsheet, and paper forms)
- How is the data to be stored and processed?
- How will the collection process be monitored to ensure integrity of the data?

Solicit identification of problems with the plan and desired improvements. Elicit the participation of the data collectors in the solution of any problems. It will provide them an opportunity to provide new ideas and insight into the development process. Support will be gained by having the parties that will be affected as active participants.

Recording procedures should be carefully considered to make them as simple as possible. Solicitation of data from project members can reduce effort and make collection more reliable.

For the failure count method, the data collection interval should be selected to correspond to the normal reporting interval of the project from which data are being collected (e.g., week, month) or an integral multiple thereof. This will facilitate obtaining data on the level of effort devoted to the software under test (person-hours and computer hours) which must be correlated with the reliability data

Step 3: Apply tools.

Availability of tools identified in the collection process must be considered. If the tools are not commercially available then time needs to be planned for their development. Furthermore, the amount of automatic data collection must be considered. To minimize the impact on the project's schedule, automated tools should be considered whenever possible.

When decisions are being made to automate the data collection process for either of the two types of data one needs to weigh certain factors. These include:

- Availability of the tool. Can it be purchased or must it be developed?
- What is the cost involved in either the purchase of the tool or its development?
- When **will** the tool be available? If it must be developed, will its development schedule coincide with the planned use?
- What impact will the data collection process have on **the** development schedule?
- Can the **tool** handle adjustments that may be needed? Can the adjustments be completed in a timely manner?
- How much overhead (people and computer time) will be needed to keep the data collection process going?

Once the tool has been developed and implemented, one needs to consider ways of ensuring the right data are being gathered. Flexibility also should be designed into the tool, as data collection requirements may change. Finally, one needs to make some type of assessment of not only what the tool saved in time and resources but also what the data collection process gained. Records could be kept of the number of faults detected after the release of the software. This could be compared with reliability estimates of similar projects that did not employ this methodology. Estimates of reduced maintenance and fault correction time could be made based upon **the** estimated current failure rate.

For the tool itself, one could estimate the amount of time and effort that would be expended if the data had been collected manually. These statistics could then yield cost estimates which would be compared with the procurement and implementation costs of the automated tool. If the cost of the automated tool is significantly higher, one certainly would question the wisdom of developing the tool. However, even if the costs come out higher, consideration must be given to future use of the tool. Once the tool has been developed it may be easily adapted over many software development efforts and could yield significant savings.

• Step 4: Provide training.

Once the tools and plans are in place, training of all concerned parties is important. The data collectors need to understand the purpose of the measurements and know explicitly what data are to **be** gathered.

• Step 5: Perform trial run.

A trial run of the data plan should be made to resolve any problems or misconceptions about the plan. This can save vast amount of time and effort when the "real thing" occurs.

• Step 6: Implement the plan.

Data must be collected and reviewed promptly. If this is not done, quality will suffer. Generate reports to show project members; they can often spot unlikely results and thus identify problems. Problems should be resolved quickly before the information required to resolve them disappears.

• Step 7: Monitor data collection.

Monitor the process as it proceeds to insure the objectives are met and the program is meeting its established reliability goals.

• Step 8: Use the data.

Don't wait to the end after the software has been released to the users to make your reliability assessments. Estimating software reliability at regular, frequent intervals will maximize visibility into the development effort, permitting managerial decisions to be

made on a regular basis.

• Step 9: Provide feedback.

This should be done as early as possible during the data collection. It is especially important to do so at the end. Those who were involved want to hear what impact their efforts had. If no feedback is given, you'll find yourself facing the problem alluded to in the beginning of this section. Namely, the parties will resist further future efforts because they see no purpose. Again, why collect data for the sake of collecting it?

# 7.2 Failure Count Data vs Execution Time Data

It is generally accepted that execution (CPU) time is superior to calendar time for software reliability measurement and modeling. If execution time is not readily available, approximations such as clock time, weighted clock time, or units that arc naturals to the applications, such as transactions, may be used [MUSA87, pp 156-158].

The following paragraphs address **failure**-count and execution time data collection to support the recommended models identified in Section 6.

#### 7.2.1 Failure-Count Data

Since the recommended models employ the number of failures detected per unit of time, these data are usually readily available. Most organizations have some type of configuration management process in place. As part of this process, a procedure for reporting failures and approving changes to the software is in place. The software problem reporting mechanism may be either manual or automatic. In addition, the problem reports may be stored within a computer data base system or a manual filing system The key is that the data can be easily extracted

Make **sure** that the problems are really software problems • some organizations use problem reporting for any type of anomaly and the time recorded on a problem report may not be the time at which the failure was experienced, it may be the time in which the

report was filled out.

Another pitfall to avoid when using problem reporting data involves forming the time intervals. Remember, the purpose is to model the number of failures detected per unit of time within a specified environment. These units should therefore be consistent in duration, manpower, and testing intensity.

Usually the information to check this is not available. All one has is data on the number of failures detected in one period or another. However, there may have been twice as many testing personnel in one period than the other. The only way to **find** out this information is to seek it out. This may involve talking with the testers or even reviewing old time sheets covering the period of interest. Generally, the longer the period of time in which the fault counts are formed the more smoothing occurs. Variations within short intervals of time will be averaged out over the longer time units.

Data may be gathered at any point within the development cycle beginning with the system test phase. Overall measurement objectives will help you determine the rate (failures reported per week, per month, or per quarter) at which data is collected. It is suggested that you start out using the number of failures reported over the shortest unit of time consistent with your objectives. If good fits are not achieved, combine intervals to the next level. For example: days to weeks, or weeks to quarters. The smoothing effect mentioned in the previous paragraph may help in the modeling process.

#### 7.2.2 Execution Time Data

This data may be collected directly or indirectly. Also, it is best to collect, when feasible, the actual execution time of a program rather than the amount of wall clock time or system active time expended. This is the actual amount of time spent by the processor in executing the instructions. Execution time gives a truer picture of the stress placed on the software. You could have large amounts of time expended on the clock but very little computations may have to be done during this period. This yields small execution

times. This would tend to give overly optimistic views of the reliability of the software. Modeling using execution time data tends to give superior results than simple elapsed wall clock time or system active time. However, the data may be difficult to collect since a monitor of the actual operating system is involved. Another source for obtaining this data is to adjust the wall clock time by a factor that represents the average computer utilization per unit of wall clock time.

If the time-between failures (wall clock or execution time) is unavailable and only grouped data (number of failures occurring per unit of time) is available, the time-between-failures can still be obtained One way is to randomly allocate the failures over the length of the time interval. Randomization will not cause errors in estimation for some of the models by more than 15 percent [MUSA87, pg.128]). A second way is the easiest to implement. Simply allocate the failures uniformly over the interval length. For example, suppose the interval is three hours in duration and three failures occurred during this period. We could then treat the time-between-failures to be each one hour in length.

Two additional considerations are: (1) adjusting the failure times to reflect an evolving program and (2) handling multiple sites / versions of the software. In the first situation, the failure intensity may be underestimated in the early stages of the program's development yielding overly optimistic views of the reliability. For the second consideration, there are multiple versions of the code being executed at different locations. In [MUSA87, pp. 162-176] both considerations are addressed.

# 7.3 Transformations Between the Two Types of Input

Programs may have the capability to estimate model parameters from either failure-count or time-between-failures data, as maximum likelihood estimation can be applied to both. However, if a program accommodates only one type of data, it is easy to transform to the other type.

If the expected input is failure-count data, it

may be obtained by transforming **time**between-failures data to cumulative time data and then simply counting the cumulative times that occur within a specified time period.

If the expected input is tune-between-failures data, convert the failure-count data by randomly selecting a number of cumulative failure times in the period equal to the count and then finding the time differences between them [MUSA87, pp. 143-146].

## 7.4 The AIAA Repository

The AIAA sponsored the development of a software reliability project repository. This repository contains data for both researchers and **practioners** alike.

### 7.4.1 Minimum Data Required

The following information represents a minimum subset of data that should be collected for any software project. It will be found useful in developing and maintaining local organization repositories as well.

# I. Project Data

The data should contain information to identify and characterize each system and effort that generates data stored in the database. Project data should allow users to categorize projects based on application type, development methodology and environment, scale, required reliability or currency. The following project-related data **are** suggested:.

- The name of each life-cycle activity (e.g., requirements definition, design, code, test, operations)
- The start and end date for each life-cycle activity
- The effort spent (in staff months) during each life-cycle activity'
- Characterize the development environment

(organic, semi-detached, or embedded)<sup>2</sup>

### II. Component Data

For each system component (e.g., subsystem, element, or module) provide the following:

- Software size in terms of executable source lines of code as well as the number of comments and the total number of object instructions
- The source language used

### III. Dynamic Failure Data

For each failure recorded the following information should be tracked:

- The activity being performed when the problem was detected (e.g., testing, operations, and maintenance)
- The date and time of the failure
- The severity of the failure (e.g., critical, major, minor)

And at least one of the following data items:

- The number of CPU hours since the last failure
- The number of runs or test cases executed since the last failure
- The number of wall clock hours since the last **failure**
- The number of test hours per test interval and number of failures detected in the interval
- Test labor hours since the last failure

#### **IV. Fault Correction Data**

For each **failure** corrected with a software fix, the following information should be **recorded:** 

<sup>1</sup> primarily required of resource modeling.

<sup>&</sup>lt;sup>2</sup> as referenced in the COCOMO framework[BOEH81]

- The date and time the **fix** was available
- · The labor hours required for correction

Also record at least one of the following data items consistent with the selected data item from the dynamic failure data list.

- . The CPU hours required for the fix
- The number of runs required to make the fix
- The wall clock hours used to make the correction

Finally, it is important to maintain corporate knowledge of the software testing and debugging effort. Therefore, have a point of contact who knows the project write down the lessons learned and have that person available to answer questions concerning the data (how they were obtained and how some of the project specific terminology translates to the current terminology).

### 7.4.2 Input for Practitioners

The above data are for use by practitioners who are interested in finding projects similar to their own projects. It also provides a guideline for defining data collection requirements when new projects are started.

## 7.4.3 Input for Researchers

In addition to the minimum data mentioned in Section 7.4.1, the AIAA Repository also contains data for research studies in software reliability measurement. These data items include:

#### I. Project Data

- Remarks about the development schedule (e.g., replans, problems, corrective actions)
- The average staff size (in staff hours) and development team experience (in years)
- The most important requirements, design, code, test, and configuration management tools and / or methods used

- The number of different organizations developing software for the project
- The Software Engineering Institute (SEI) index of the development environment and the assessment method
- The most important tool and model used for software reliability estimation

### II. Component Data

- The name and model of the development and target hardware
- Average and peak computer resource utilization (e.g., CPU busy, memory utilization, and input / output channel utilization)

### III. Dynamic Failure Data

- The type of the failure (e.g., interface, syntax)
- The method of fault / failure detection (e.g., inspection, system abort, invalid output)
- The unit complexity (e.g., McCabe Cyclomatic) and size where the fault was detected

#### **IV. Fault Correction Data**

 The type of fix (e.g., software change, documentation change, requirements change, no change)

#### 8.0 BIBLIOGRAPHY

- [ABDE86] Abdel-Ghaly, A. A., Chan, P. Y., and Littlewood, B., "Evaluation of Competing Software Reliability Predictions," *IEEE Transactions on Soft*ware Engineering, SE-12, 9, pp 950-967, 1986.
- [BOEH81] Boehm, B. W., Software Engineering Economics, Prentice-Hall, New York, 1981.
- 3. [BOWE87] Bowen, John B., "Application of a Multi-Model Aproach

- to Estimating Residual Software Faults and Time Between Failures," *Quality and Reliability Engineering International*, Vol. 3.41-51 (1987) pp. 41-51.
- **4.** [BROC92] Brocklehurst, S. and Littlewood, B., "New Ways to Get Reliability Measures," *IEEE Software*, July 1992, pp. 34-42.
- [BROO80] Brooks, W.D. and Motley, R.W., Analysis of Discrete Software Reliability Models, Technical Report #RADC-TR-80-84, Rome Air Development Center, 1980
- 6. [CROW77] Crow, L., Confidence Interval Procedures for Reliability Growth Analysis, Technical Report #197, U.S. Army Material Systems Analysis Activity, Aberdeen Proving Grounds, Maryland.
- 7. [DUAN64] Duane, J.T., "Learning Curve Approach to Reliability Monitoring," *IEEE Transactions on Aerospace*, Volume 2, pp. 563-566.
- [FARR83] Farr, W. H., A Survey of Software Reliability Modeling and Estimation, Technical Report #82-171, Naval Surface Warfare Center, Dahlgren, Virginia.
- [FARR91] Farr, W. H. and Smith, O. D., Statistical Modeling and Estimation of Reliability Functions for Software (SMERFS) Users Guide, NAVSWC TR-84-373, Revision 2, Naval Surface Warfare Center, Dahlgren, Virginia.
- [FREE88] Freedman, R. S. and M. L. Shooman, An Expert System for Software Component Testing, Final Report, New York State Research and Development Grant Program, Contract No. SSF(87)-18, Polytechnic University, Oct. 1988.
- 11. **[GIFF84]** Gifford, D., and Spector, A., "The TWA Reservation System," *Communications of the ACM*, pp. 650-665, Vo12, No. 27, July 1984.

- 12. **[GOEL79]** Goel, A. and Okumoto, K., "Time-Dependent Error-Detection Rate for Software Reliability and Other Performance Measures," *IEEE Transactions on Reliability*, Vol. R-28, No. 3, pp. **206-211.**
- 13. **[HECH86a]** Hecht, H. and Hecht, M., "Software Reliability in the System Context," *IEEE Transactions on Software Engineering*, January 1986.
- 14. **[HECH86b]** Hecht, H. and Hecht, M., "Fault Tolerant Software," in *Fault Tolerant Computing*, D. K. Pradhan, ed., Prentice Hall, 1986.
- 15. **[HECH89]** Hecht, H., "Talk on Software Reliability," given at AIAA Software Reliability Committee Meeting, Colorado Springs, CO., August 22-25, 1989.
- [HOEL71] Hoel, P. G., Introduction to Mathematical Statistics, Fourth Edition, John Wiley & Sons, New York, NY, 1971.
- 17. **[IYER83]** Iyer, R. K., and Velardi, P., **A Statistical Study of Hardware Related Software Errors in** MVS, Stanford University Center for Reliable Computing, October 1983.
- 18. **[JELI72]** Jelinski, **Z.** and P. Moranda, "Software Reliability Research," in W. Freiberger, ed., **Statistical Computer Performance Evaluation**, Academic Press, New York, NY, 1972, pp. 465-484.
- 19. **[JOE85]** Joe, H. and Reid, N., "Estimating the Number of Faults in a System," *Journal of the American Statistical Association*, 80(389), pp. 222-226.
- 20. [KAFU87] Kafura, D. and Yerneni, A., Reliability Modeling Using Complexity Metrics, Virginia Tech University Technical Report, Blacksburg, VA, 1987
- 21. **[KANO87]** Kanoun, K. and Sabourin,

- T., "Software Dependability of a Telephone Switching System," *Proc.* 17th IEEE Int. Symposium on Fault-Tolerant Computing (FTCS-17), Pittsburgh, PA, 1987.
- 22. **[KARU92]** Karunantithi, N., Whitely, D., and Malaiya, Y. K., "Using Neural Networks in Reliability Prediction," *IEEE Software*, July 1992, pp. 53-60.
- 23. [KHOS91] Munson, J. C. and Khosh-goftaar, T. M., "The Use of Software Complexity Metrics in Software Reliability Modeling," Proceedings of the International Symposium on Software Reliability Engineering, Austin, TX, May 1991, pp. 2-11.
- 24. [KLIN80] Kline, M. B., "Software & Hardware R&M: What are the Differences?" Proceedings Annual Reliability and Maintainability Symposium, 1980, pp. 179-185.
- 25. **[LAPR84]** Laprie, J. C., "Dependability Evaluation of Software Systems in Operation," *IEEE Trans. on Software Eng.*, Vol. SE-10, Nov 84, pp 701-714.
- 26. [LIPO86] Lipow, M. and Shooman, M. L., "Software Reliability," in Consolidated Lecture Notes Tutorial Sessions Topics in Reliability & Maintainability & Statistics, Annual Reliability and Maintainability Symposium, 1986.
- 27. [LITT73] Littlewood, B. and Verrall, J. L., (June 1974, "A Bayesian Reliability Model with a Stochastically Monotone Failure Rate," *IEEE Transactions on Reliability*, pp. 108- 114.
- 28. **[LITT79]** Littlewood, B. "Software Reliability Model for Modular Program Structure," *IEEE Trans. on Reliability*, R-28, pp. 241-246, Aug. 1979.
- 29. **[LITT86]** Littlewood, B., Ghaly, A., and **Chan**, P. Y., "Tools for the Analysis of the Accuracy of Software Reliability Predictions", (Skwirzynski, J. K., Editor), **Software System Design Methods**, NATO AS1 Series, F22, **Springer-**

- Verlag, Heidleberg, pp. 299-335.
- 30. **[LITT90]** Fenton, N. and Littlewood, B., "Limits to Evaluation of Software Dependability," *Software Reliability and Metrics*, Elsevier Applied Science, London, pp. 81-110.
- 3 1. **[LLOY77]** Lloyd, D. K. and Lipow, M. *Reliability: Management, Methods, and Mathematics,* 2nd Edition, 1977, ASQC.
- 32. **[LU92]** Lu, M., Brocklehurst, S., nd Littlewood, B., "Combination of Predictions Obtained from Different Software Reliability Growth Models," *Proceedings of the Tenth Annual Software Reliability Symposium,* Denver, CO, June 1992, pp. 24-33.
- 33. **[LYU92]** Lyu, M. R. and Nikora, A., "Applying Reliability Models More Effectively," *IEEE Software*, July 1992, pp. 43-52.
- 34. [MATH92] Mathur, A. P., and Horgan J. R., "Experience in Using Three Testing Tools for Research and Education in Software Engineering," Proceedings of the Symposium on Assessment of Quality Software Development Tools, pp. 128-143, May 27-29, 1992, New Orleans, LA
- 35. [MCCA87] McCall, J. A., et al, "Methodology for Software and System Reliability Prediction," Final Technical Report, Prepared for RADC, Science Applications International Corporation, November 1987, RADC-TR-87- 17 1.
- 36. [MELL86] Mellor, P., "State of the Art Report on Software Reliability." *Infotech,* London, 1986.
- 37. [MIL-HDBK-217E] Military Handbook Reliability Prediction of Electronic Equipment, MIL-HDBK-217E, Rome Air Development Center, Griffis AFB, NY 13441-5700, Oct. 27, 1986. Naval Publications and Forms Center, Code 3015, 5801 Tabor Ave., Philadelphia, PA 19120.

- 38. **[MUSA75]** Musa, J., "A Theory of Software Reliability and Its Application," *IEEE Trans. Software Eng.*, Vol. **SE-1**, No. 3, September 1975, pp 312-327.
- 39. **[MUSA84]** Musa, J.D., Okumoto, K., "A Logarithmic Poisson Execution Time Model for Software Reliability Measurement," *Proceedings Seventh International Conference on Software Engineering,* Orlando, pp. 230-238.
- 40. [MUSA87] Musa, J. D., Iannino, A., and Okumoto, K., Software Reliability: Measurement, Prediction, Application. McGraw-Hill, New York, 1987.
- 41. **[MUSA92]** Musa, J. D., \*'Determining the Operational Profile," available from the author.
- 42. **[NPRD85]** Nonelectronic Parts Reliability Data, NPRD-3, Reliability Analysis Center, Rome Air Development Center, Griffis AFB, NY 134415700, 1985, NTIS ADA163514
- 43. **[ROOK91]** Rook, P. *Software Reliability Handbook*, Elsevier Applied Science, London, 1990.
- 44. **[SCHN75]** Schneidewind, N. F., "Analysis of Error Processes in Computer Software," Proceedings of the International Conference on Reliable Software, **IEEE** Computer Society, 21-23 April 1975, pp. 337-346.
- 45. **[SCHN92]** Schneidewind N. F. and Keller, T. M., "Applying Reliability Models to the Space Shuttle," *IEEE Software*, July 1992, pp. 28 33.
- 46. **[SHOO76]** Shooman, M. L., "Structural Models for Software Reliability Prediction," Second National Conf. on Software Reliability, San Francisco, CA, October 1976.
- **47.** [SHOO83] Shooman, M. L., Software Engineering: Design, Reliability, and Management, McGraw-Hill Book Co, New York, NY, 1983.

- 48. [SHOO90a] Shooman, M. L., Probabilistic Reliability: An Engineering Approach, McGraw-Hill Book Co., New York, NY, 1968, 2nd. Edition, Krieger, Melbourne, FL, 1990.
- 49. **[SHOO90b]** Shooman, M. L., "Early Software Reliability Predictions," Software Reliability Newsletter, Technical Issues Contributions, IEEE Computer Society Committee on Software Engineering, Software Reliability Subcommittee, **9/1 1/90.**
- 50. **[SIEF89]** Siefert, D. M., (March **1989)**, "Implementing Software Reliability Measures," *The NCR Journal*, Vol. 3, No. 1, pp. 24-34.
- 5 1. **[STAR92]** Stark, G. E., "Software Reliability Measurement for Flight Crew Training Simulators," *AIAA Journal of Aircraft*, Vol. 29, No. 3, May-June 1992, pp. 355-359.
- 52. [TAKA85] Takahashi, N. and Kamayachi, Y., "Am Empirical Study of a Model for Program Error Prediction," Proceedings 8th International Conference on Software Engineering, London, pp. 330-336.
- 53. **[YAMA83] Yamada,** S., Ohba, M., and Osaki, S., "S-Shaped Reliability Growth Modeling for Software Error Detection," *IEEE Transactions on Reliability,* Vol. R-32, No. 5, pp. **475-**

# **APPENDIX A**

# ADDITIONAL SOFTWARE RELIABILITY ESTIMATION MODELS

This appendix contains descriptions of four additional models available to a researcher or software reliability analyst for use on projects, that were not discussed in Section 6 of the recommended practice. These models may be useful on projects where the assumptions of the models recommended in section six do not apply or the models in section six do not closely *fit* the data. It is recommended to use more than one model in practice since the computation time for the analysis of multiple models is reasonable.

### A.1 Duane's Model

# A. 1.1 Duane's Model Objectives

This model assumes that we are dealing with the times of failures occurrences. number of such occurrences considered per unit of time is assumed to follow a nonhomogenous Poisson process. model was originally proposed by J. T. Duane who observed that the cumulative failure rate when plotted against the total testing time on log-log paper tended to follow a straight line. This model has had some success in its application [DUAN64]. It is best applied later in the testing phase or beyond. The cummulative operation of summing the total number of errors to date tends to have a smoothing effect and hence promotes the linear relation present in the model.

#### A. 1.2 Duane's Model Assumptions

The specific assumptions are:

- The software is operated in a similar operational profile as the anticipated usage.
- The failure occurrences are independent.

• The cumulative number of failures at any time t, [N(t)], follows a Poisson distribution with mean m(t). This mean is taken to be of the form  $m(t) = \lambda t^b$ .

### A. 1.3 Duane's Model Structure

If  $\frac{\mathbf{m(t)}}{t} = \frac{\lambda t^b}{t}$  is plotted on log-log paper a straight line-of the form  $Y = \mathbf{a} + bX$  with  $\mathbf{a} = \ln \lambda$ , b = b, and  $X = \ln(t)$  is obtained.

Maximum likelihood estimates are shown by [CROW77] to be:

$$\hat{\lambda} = \frac{n}{t_n^{\hat{b}}}$$

$$\hat{\mathbf{b}} = \frac{\mathbf{n}}{\sum_{i=1}^{n-1} \ln(\mathbf{t}_n / \mathbf{t}_i)}$$

where the  $t_i$ 's are the observed failure times in either CPU time or wall clock time and n is the number of failures observed to date.

Least Squares estimates for **a** and **b** of the straight line (see previous Structure section) on log-log paper can be derived using standard linear regression estimates.

# A. 1.4 Duane's Model Data Requirements

The model requires the time of the failure occurrences, i.e.  $t_{i_1}$ ,  $i = 1, \ldots, n$ .

# A.2 Brooks and Motley's IBM Model

# A .2.1 Brooks and Motley's Model Objectives

This model attempts to account for the fact that the software may be developed incrementally so that all of the modules may not be under test at the same time. Additionally, the amount of the program under test could require different expenditures of resources (e.g., staff-hours or

CPU-hours expended). This model was designed to handle these situations and can therefore be applied at either the system or module level. For consistency with the other models in this recommended practice, only the system level model and parameter estimates are presented here. See [FARR83] for a general treatment of these model variations.

# A.2.2 Brooks and Motley's Model Assumptions

The number of faults detected in a given test period **i** follows either a Poisson distribution or a binomial distribution. Specifically the assumptions are:

- The number of software faults detected on each test occasion is proportional to the number of faults at risk for detection which is proportional to the remaining number of faults.
- This proportionality factor or probability (denoted as q for the binomial model and φ for the Poisson) of detecting any fault during a specified unit interval of testing is constant over all test occasions and independent of fault detection.
- The faults reintroduced in the correction process are proportional to the number of faults detected.

One consideration when using this model is the second assumption of a constant fault detection probability. If this probability is changing drastically over time another of the models considered in this document may be more appropriate. The fluctuation in **the** fault detection probability can sometimes be seen in the initial testing phase as the testers are learning the system. The S-shaped model discussed in the next section may be more appropriate in this case.

# A. 2.3 Brooks and Motley's Model Structure

#### **Binomial Model**

Suppose  $J_i$  is the index set of those modules tested on occasion i, N is the total number of faults in the software program at the beginning of testing,  $w_i$  is the weight assigned to module j, q is the error detection probability given in the second assumption above and a is the probability of correcting a fault in the software without introducing new faults. Then the binomial model over the ith test occasion (i = 1,...,K) can be shown to be [BROO80]:

$$P(X = n_i) = {\overline{N}_i \choose n_i} q_i^{n_i} (1 - q_i)^{\overline{N}_i - n_i}$$

where

 $\bar{N}_i$  = the number of faults remaining and subject to detection at the start of the ith test occasion

$$= \sum_{j \in J_i} (w_j N - a N_{i-1,j})$$

 $\mathbf{q_i} = [1 - (1 - \mathbf{q})^{\mathbf{K_i}}]$  where  $\mathbf{K_i}$  is the system test effort on the  $i^{\text{th}}$  test occasion

probability of fault detection in the *ith* test occasion

and

$$n_i = \sum_{j}^{J_i} n_{ij}$$

total number of faults found in the *ith* test occasion over the modules being tested.

For the Binomial model the maximum likelihood estimates of the parameters can be shown to solve the following system of equations:

$$0 = \sum_{i=1}^{K} \left( \ln \left[ \frac{\overline{N}_i}{\overline{N}_i - n_i} \right] + K_i \ln(1 - q) \right) \sum_{j \in J_i} w_j$$

$$0 = \sum_{i=1}^{K} \left( \ln \left[ \frac{\overline{N}_i}{\overline{N}_i - n_i} \right] + K_i \ln(1 - q) \right) \sum_{j \in J_i} N_{i-1, j}$$

$$0 = \sum_{i=1}^{K} \left( \frac{n_i K_i}{1 - (1 - q) K_i} \right) - K_i \overline{N}_i$$

### **Poisson Model**

Suppose  $J_i$  is the index set of those modules tested on occasion i, N is the total number of faults in the software program at the beginning of testing,  $w_j$  is the weight assigned to module j,  $\phi$  is the error detection probability given in the second assumption above for the Poisson and  $\alpha$  is the probability of correcting a fault in the software without introducing new faults. Then the Poisson model over the  $i^{th}$  test occasion (i = 1,...,K) of length  $t_i$  can be shown to be [FARR83]:

$$P(X = n_i) = \frac{\left(\overline{N}_i \phi_i\right)^{n_i} e^{-\overline{N}_i \phi_i}}{n_i!}$$

where

 $\bar{N}_i$  = the number of faults remaining and subject to detection at the start of the  $i^{th}$  test occasion

$$= \sum_{j \in J_i} (w_j N - \alpha N_{i-1,j})$$

$$f_i \quad = \quad \left[1-\left(1-\varphi\right)^{t_i}\right]$$

= probability of fault detection in the *ith* test occasion where **t**<sub>i</sub> is the total time spent for the *ith* test occasion

and

$$n_i = \bar{N}_i f_i$$

= total expected number of faults

The maximum likelihood estimates for the Poisson model parameters N,  $\phi$  and a are found as the solution of the following three equations:

$$0 = \sum_{i=1}^K \left( \sum_{j \in J_i} w_j \right) \left( \frac{n_i}{\overline{N}_i} - \phi_i \right)$$

$$0 = \sum_{i=1}^{K} \left( \sum_{j \in J_i} N_{i-1, j} \right) \left( \frac{n_i}{\overline{N}_i} - \phi_i \right)$$

$$0 = \sum_{i=1}^{K} t_{i} (1 - \phi)^{t_{i}} \left[ \frac{n_{i}}{1 - (1 - \phi_{i})^{t_{i}}} - \overline{N}_{i} \right]$$

For both the binomial model and the Poisson it is best to **fix** the value for a (the probability of correcting faults in the code without introducing new ones), as the three simultaneous equations are extremely difficult to find the solutions for. If a is fixed the three equations in both cases reduce to two equations with the last equation in each set disappearing. Brooks and Motley suggest choosing values for a ranging from 0.85 to 1.00.

# A.2.4 Brooks and Motley's Model Data Requirements

The data required to implement either of these two model forms are:

- The length  $t_i$  of the  $i^{th}$  test occasion.
- The total number of faults, (ni), found in the i<sup>th</sup> test occasion over the modules being tested.
- The modules under test during the *i*<sup>th</sup> test occasion.
- The probability of correcting faults in the code without introducing new ones, a.

# A.3 Yamada, Ohba, and Osaki's S-shaped Reliability Growth Model

## A. 3.1 S-Shaped Reliability Growth **Model Objectives**

This model assumes that we are dealing with the times of failures occurrences. The number of such occurrences considered per unit of time is assumed to follow a nonhomogeneous Poisson process. This model was proposed by **Yamada**, Ohba, and Osaki [YAMA83]. It is based upon Goel and Okumoto's Nonhomogeneous Poisson Process (NHPP) [GOEL79]. The difference is that the mean value function of the Poisson process is s-shaped in nature to allow for a learning curve effect. At the beginning of the testing phase the fault detection rate is relatively flat but then increases exponentially as the testers become familiar with the program Finally it levels off near the end of testing as faults become more difficult to uncover. This behavior is best fitted by an Sshaped **model**; hence the basis of their model.

# A, 3.2 S-Shaped Reliability Growth **Model Assumptions**

The basic assumptions are:

- The software is operated in a similar operational profile as the anticipated
- The failure occurrences are independent and random
- · The initial fault content is a random variable.
- The time between failures (i 1) and **i** depends on the time to failure (i - 1).
- Each time a failure occurs, the fault which caused it is immediately removed, and no other faults are introduced.

### A. 3.3 S-Shaped Reliability Growth Model Structure

The specific model is:

 $P(N_t = n)$  = probability that the cumulative number of faults up to time t,  $N_t$ , is equal to n

= 
$$\frac{M(t)^{n} \exp(-M(t))}{n!}$$
 where n = 0, 1,...

with

the mean value function for the M(t) =Poisson process

= 
$$a(1 - (1 + bt)e^{-bt})$$
 with both a, b > 0

and with initial conditions

$$\mathbf{M}(0) = 0$$

$$\mathbf{M}(\infty) = a$$

The fault detection rate is therefore:

$$\frac{dM(t)}{dt} = ab^2 te^{-bt}$$

Letting  $\mathbf{n_i}$ , i = 1,...,k be the cumulative number of faults found up to time  $t_i$ , i =k, the maximum likelihood estimates for a and b are shown to satisfy the following pair of equations.

$$\hat{\mathbf{a}} = \frac{\mathbf{n_k}}{\left(1 - \left(1 + \hat{\mathbf{b}}\mathbf{t_k}\right)\mathbf{e}^{-\hat{\mathbf{b}}\mathbf{t_k}}\right)}$$

and

$$\hat{a}t_{k}^{2}e^{-\hat{b}t_{k}} = \sum_{i=1}^{k} \left[ \frac{(n_{i} - n_{i-1})(t_{i}e^{-\hat{b}t_{i}} - t_{i-1}e^{-\hat{b}t_{i-1}})}{(1 + \hat{b}t_{i-1})e^{-\hat{b}t_{i-1}} - (1 + \hat{b}t_{i})e^{-\hat{b}t_{i}}} \right]$$

This model does an excellent job in both fitting and given set of data and for prediction when this s-shaped phenomenon is observed.

# A. 3.4 S-Shaped Reliability Growth Model Data Requirements

The model requires the failure times  $t_i$ , i = 1, . . . . k as input data.

# A. 4 Jelinski / Moranda Reliability Growth Model

## A. 4.1 Jelinski / Moranda Model Objectives

The basic idea behind this model is that failure occurrence rate is proportional to the number of faults remaining, the rate remains constant between failure detections and the rate is reduced by the same amount after each fault is removed. The last idea means that the correction of each fault has the same effect in reducing the hazard rate of the program.

# A.4.2 Jelinski / Moranda Assumptions

The basic assumptions of the Jelinski-Moranda Model are:

- The rate of failure detection is proportional to the current fault content of a program.
- All failures are equally likely to occur and are independent of each other.
- Each failure is of the same order of severity as any other failure.
- The failure rate remains constant over the interval between failure occurrences.
- The software is operated in a similar manner as the anticipated operational usage.
- The faults are corrected instantaneously without introduction of new faults.

### A .4.3 Jelinski / Moranda Structure

Using these assumptions the hazard rate is defined as:

$$z(t) = \boldsymbol{\phi[N-(i-1)]}$$

where t is any point between the discovery of the (i-1)th failure and the ith failure. The quantity  $\phi$  is the proportionality constant given in the first assumption. N is the total number of faults initially in the program. Hence if (i-1) faults have been discovered by time t, there are N - (i-1) remaining faults. The hazard rate is proportional to this remaining number. As a fault is discovered the hazard rate is reduced by the same amount,  $\phi$ , each time.

If  $X_i = t_i - t_{i-1}$ , i.e. the time between the discovery of the **ith** and the (i-1)st fault for i = 1,...,n where  $t_0 = 0$ ; using the fourth assumption, the  $X_i$ 's are assumed to have an exponential distribution with rate  $z(t_i)$ . That is:

$$f(X_i) = f[N - (i - 1)] \exp(-f[N - (i - 1)]X_i)$$

This leads to the maximum likelihood estimates of  $\phi$  and N as the solutions to the following two equations:

$$\hat{\phi} = \frac{n}{\hat{N}\left(\sum_{i=1}^{n} X_i\right) - \sum_{i=1}^{n} (i-1)X_i}$$

$$\sum_{i=1}^{n} \frac{1}{\hat{N} - (i-1)} = \frac{n}{\hat{N} - \frac{1}{\sum_{i=1}^{n} X_{i}}} \left( \sum_{i=1}^{n} (i-1) X_{i} \right)$$

# A.4.4 Jelinski / Moranda Data Requirements

The model may use either of the following items as input data for the parameter estimation:

- The time between failure occurrences,
   i.e., the X<sub>i</sub>'s.
- The total duration of the failure occurrences, i.e.  $t_i = \sum_{j=1}^{i} X_j$

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ANSI/AIAA R-013-1992

# APPENDIX B

# AUTOMATED SOFTWARE RELIABILITY MEASUREMENT TOOLS

This appendix provides a list of the known software reliability measurement tools available to practitioners and researchers. It is summarized from an AIAA special report, An Evaluation of Tools for Modeling Software Reliability, and contains only those tools with survey information available at the time of publication. It should be noted that additional tools are arriving on the market and these tables represent the status as of this printing.

	Supplier and	Naval Surface Warfare	Reliability and Statis-	Data & Analysis Center
	contact	Center (NSWC/DD)	tical Consultants, Ltd	for Software (DACS)
		Dr. William Farr NSWCDD Dahlgren, VA 22448-5000 (703) 663-4719	Dr. Bev Littlewood Center for Software Reliability Northampton Sq. London EC1 <b>V0HB</b> , England (+44 71 477 8420) (+44 71 477 8585) FAX	DACS RDD/COED Griffiss AFB, NY 13441 (315) 336-0937
	Tool Name	Statistical Modeling and Estimation of Reliability Functions for Software (SMERFS)	Software Reliability Modeling Programs (SRMP)	GOEL
	Models	Littlewood/Verrall Musa Basic Musa/Okumoto Geometric Execution Time NHPP Generalized Poisson NHPP Brooks/Motley Schneidewind S-Shaped	Musa/Okumoto Duane Jelinski/Moranda (JM) Goel/Okumoto Bayesian JM Littlewood/Verrall Littlewood Keiller/Littlewood Littlewood NHPP	Goel/Okumoto
	Hardware	Cyber 170/760, DEC VAX, IBM PC (some versions require a math coprocessor)	Sun Microsystem Work- station or IBM PC com- patible with a math coprocessor	<b>IBM PC</b>
	Minimum Operating System	DEC VMS, MS DOS 3.0, <b>Cyber</b> Operating System	MS DOS 3.0	MS DOS 2.11
	Minimum Memory	256 K	500 K	256K
	Current Version	4.0	1.0	1.0
Release Data	Version Release Date	Jun-90	May-88	Nov-87
	Original Date Distributed copies	Oct-83 300	May-88 Unknown	Nov-87 68
	Development Language	FORTRAN '77	FORTRAN	Unknown
Program De- veloped for	Commercial Use Project Specific Use	х	Х	Х
Program Structure	Menu-Driven Command- Driven	x	Х	Х
	Integrated System Stand Alone	Х	Х	Х
	Tool Cost (\$)		\$5,000	\$50

	Supplier and Contact	AT&T Bell Laboratories	AT&T <b>Bell</b> Laboratories	Software Quality Tools
		Dr. William Everett AT&T Bell L&oratories Rm 2L- 503 Crawfords Corner Road Holmdel. NJ 07733-(908) 949 2334	Dr. William Everett AT&T Bell Laboratories Rm 2L- 503 Crawfords Come-r Road Hohndel, NJ 07733-(908) 949 2334	Thomas L. Wilson Software Quality Tools 2000 West Park Drive Suite 200 Westborough, MA 01581 (508) 366-5045
	Tool Name	Program for Software Reliability and System Test Schedule Estimation	RELTOOLS (a PC-based version called SRE tools is currently in beta test)	Software Quality Management System (SQMS)
	Models	Musa Basic  Musa/Okumoto	Musa Basic Musa/Okumoto	Musa Basic
	Hardware	<b>CDC 6000/7000, IBM 360/370,</b> DEC VAX, Univac 1100. Honeywell 6000	Any platform running UNIX system V operating system (IBM PC for SRE tools)	Sun SPARC station
	Minimum Operating System	Unknown	See above	sun OS 4.1,Open Windows 2.0
	Minimum Memory Requirements	100 K	100 K	8 Meg
	Current Version	1.0	2.0	1.2
Release Data	Current Version Released	1977	1988	Mar-91
	Original Version Released	1977	Sept-87	Ott-90
	Distributed coples	>100	13	>10
	Developmenl Language	FORTRAN	FORTRAN '77	С
Program Developed for	Commercial Use		X	X
	Project Specific Use	X		
Program Structure	Menu-Driven Command- Driven Integrated	X	X	
	System Standalone Tool	X	X	
	Cost (\$)	Public Domain	\$300	\$25,000

	I		ann a	1400
	Supplier and	AT&T Bell Laboratories	CEP-Systemes	MBB
	Contact			Deutsche Aerospace
				Munich
		Dr. William Everett AT&T	M. Sulvoin Mary CED	Mr. R. Borcz
			Mr. Sylvain Metge CEP-	
		Bell Laboratories Rm 2L-	Systemes	MBB Deutsche Aerosp.
		503	150 me Vauquelin	space Comm. & Propul.
		Crawfords Comer Road	<b>Immeuble</b> Europolis Bat.	System Div. Mail Code
		Hohndel. NJ 07733 (908)	A	KQ114
		949 2334	31081 Toulouse Cedex	<b>D8000</b> Munich 80
		949 2334		
			France	Germany
	Tool Name	SRE Toolkit	SoRel	SOFTREI
	Models	Musa Basic	4 models implemented	Shooman
		Musa/Okumoto		
	Hardware	Any platform running	Macintosh II with a math	IBM PC
	Haluwaic			IBM TC
		Unix System V or	coprocessor	
	<b></b>	MS/DOS		
	Minimum	see above	Macintosh	MS DOS
	Operating			
	System			
	Minimum	120K	200K	256K
		1201	2001	
	Memory			
	Requirements		<u> </u>	
	Current	1.0	1.0	2.0
	Version			
Release Data	Current	May-91	May-91	1989
Hereuse Butu	Version	•	·	
	Released	37 4004	34 . 01	1990
	Original	May 1991	May-91	1990
	Version			
	Released		-	
	Distributed	> 200	Unknown	1
	copies	- 200		_
		С	Pascal	Pascal
	Development		rascai	i ascai
	Language			
Program	Commercial	Х	X (all documentation in	
Developed	Use		French)	[
for	1			
	Project			x
	Specific Use		1	
			v	Х
	Menu-Driven		X X	Α
Program	Command-	Х	X	
Structure	Driven			
	Integrated			1
	System			
		Х	X (needs Excel <sup>TM</sup> for	Х
	Standalone	^	-	1 ^
	Tool		plotting)	<del> </del>
	Cost (\$)	Free with attendance at 3-	30,000FF(3,000FF for a	Unknown
		day training seminar	non-profit organization)	

# **APPENDIX C**

# DETERMINING SYSTEM RELIABILITY

Reliability analysis involves approximations, assumptions, and often the use of generic rather than field specific data. estimates are often off by a factor of 1.5 or 2. For example, suppose that the system requirements call for a hardware MTTF of 1,000 hours and a software MTTF of 1,000 hours (yielding a system MTTF of 500 hours). Conservative design procedures would be to design for 2,000 hours MTTF for both the hardware and software, so that even if reality is worse than the model assumptions, there is a built in safety factor Thus, neither hardware nor of 100%. software models need give exact predictions to be important analysis techniques. [SHOO90a].

This appendix describes methods for combining hardware and software reliability predictions into a system prediction.

# C. 1 Predict Reliability for Systems Composed of (Hardware and Software) Subsystems

A simple way of dealing with the reliability of a system composed of hardware and software is to make a structural model for the system. The most common types of structural models in use are reliability block diagrams (reliability graphs) and reliability fault trees. If the hardware and software modes of failure are independent, then the system reliability, Rs, can be treated as the product of the hardware and software reliability, and a separate model can be made for the hardware and software. Consider the following example:

A railroad boxcar will be automatically identified by scanning its serial number (written in bar code form) as the car rolls past a major station on a railroad system. Software compares the number read with a data base for match, no match, or partial match. A simplified hardware graph for the system is given in Figure C.1, and the hardware reliability, R(HW), in Equation (C.1).

$$R(HW) = RS * RC * RD * RP \qquad (C.1)$$

The software graph is shown in Figure C.2, the software reliability, **R(SW)**, in Equation **(C.2)**, and combining these equations, the system reliability R(SYSTEM) is given in **(C.3)**.

$$R(SW) = RF * RL * RD * RA \qquad (C.2)$$

$$R(SYSTEM) = R(HW) * R(SW)$$
 (C.3)

In a more complex case the hardware and software are not independent and a more complex model is needed. For example consider a fault tolerant computer system with the computers, Cl, C2, C3, the same software on each computer (SW1, SW2, SW3), and an output majority voter (answer is the majority output) [SHOO90], Appendix H]. Since the software [SW] is the same

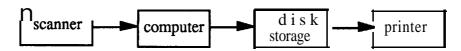


Figure C. 1 The Hardware Model of a Railroad Boxcar Identification System



Figure C.2 The Software Model of a Railroad Boxcar Identification System

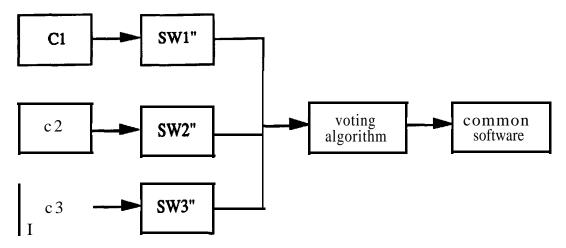


Figure C.3 A Reliability Graph for a Fault Tolerant Computer System

Note that such models work best at a high level where there will be a modest number of subsystems.

most of the failures [say 90%] due to design faults, errors in specification, etc. are common to all processors and appear in series in a graph model. Since the computers do not have the same internal state, say 10% of the software failures (SW") are independent as is shown in Figure C.3 and the reliability equation, Equation (C.4), is written in terms of the subsystem probabilities, Pr.

$$R = Pr [(C1 * SW1" + C2 * SW2" + c3 SW3") * v * SW']$$
(C.4)

The hardware reliabilities for such system models are derived **from** test and operational data on the number of equipment failures for each subsystem and the total number of hours of test. Similarly one would take data on system failures traceable to the software, however, one would need to count these failures as system level failures to use these models. This is the reason why such models can not be applied at too low a level.

More detailed micro models have been formulated and described in the literature, however while they appear theoretically sound, unlike the macro models previously described, they have not been applied to actual projects as yet. These micro models which have been developed [SHO076, LLOY77, LITT79, LAPR84, HECH89] focus on a simple representation of the

software structure. As an example consider the following model based on representing the software by a structure with i major paths [SHOO83, FREE88, SHOO90b]

During operation (execution) of the software, each of these paths is selected with frequency  $\mathbf{f_i}$ , and the execution time of each case is  $\mathbf{t_i}$ . There is a certain probability that in executing case i, a residual software error will be encountered which results in system failure. This failure probability is denoted by  $\mathbf{q_i}$ .

Development of the model **[(SHOO76,** SHOO83) pp. **378-384]** leads to an expression for the system failure rate which depends on the  $\mathbf{f_i}$ ,  $\mathbf{t_i}$ , and  $\mathbf{q_i}$  parameters:

$$\mathbf{Z(0)} = \frac{\sum_{j=1}^{i} f_{j} q_{j}}{\sum_{j=1}^{i} f_{j} (1 - \frac{q_{i}}{2}) t_{j}'}$$
(C.5)

Note that the symbol for executed time of path i has been given a prime,  $t'_j$ , to differentiate it **from** the system operating time t.

If the  $q_i$  values are small, as they may be in most cases, then Equation **(C.5)** simplifies to:

$$z(0) = \frac{\sum_{j=1}^{i} f_{j} q_{j}}{\sum_{j=1}^{i} f_{j} t_{j}'}$$
 (C.6)

We can interpret Eq. (C.6) in a simple fashion. The failure rate **z(0)** is just the ratio of the weighted failure probabilities and the weighted running times (to failure or success), yielding failures per hour.

Note that the failure rate function **z(0)** in Equation (C.6) is independent of operating time t. Thus, substituting **z(0)** into the standard reliability expression [SHOO90a], yields:

$$R(t) = exp[\int_{0}^{t} -z(0)dx] = exp(-z(0)t)$$
 (C.7)

The mean time to failure, MTTF, is given by:

$$MTTF = \int_{0}^{\infty} R(t)dt$$
 (C.8)

Since **z(0)** is independent oft, substitution of (C.7) into **(C.8)** yields

$$MTTF = \frac{1}{z(0)}$$
 (C.9)

# C .2 Predict Reliability in the Engineering Phase

### C .2.1 Software System

Software reliability prediction in engineering phase is basically the same as measuring the software reliability in testing and operational phase. Since during the engineering phase, the software is only considered as part of the whole system (usually represented by a few blocks in the overall system block diagram), software reliability usually affects system reliability partially. However, in the case where software is involved in the operation of a critical section of the system, reliability of that software portion will have immediate

impact on the system reliability. Therefore, it is important to separate the reliability prediction for critical software portions from that for noncritical portions.

Another important issue of predicting software reliability in the engineering phase is to correctly identify the expected operational profile, especially when the functionality of the software depends on certain assumptions made by the hardware, and made by the interfaces in between hardware and software. There might be some discrepancies which will not be caught by the software integration testing, and would have to be resolved in the system engineering phase.

# C.2.2 Systems Composed of Hardware and Software Subsystems

In concept this is essentially the same task as that discussed in Section 5 for the Test and Operational Phases; however, during the System Engineering Phase, we do not have operational or test data for our current project. We must rely on historic data recorded in raw form or distilled into a reliability estimate for the hardware and software within the system. In the past the field of hardware reliability has been quite successful in collecting, analyzing, and recording field failure data for failure rate estimates of various component reliabilities. (The two best known hardware reliability manuals are [MIL-HDBK-217E] and [NPRD85]).

Such values are generally used for estimating the reliability of new hardware locating similar components or equipments in the historical data base. One of the objectives of this AIAA project is to evolve such a data base.

In the interim and even after such a data base is established, one will often need to scale historical data to adjust for more logical and thorough development procedures. The following technique can be helpful in this regard [SHOO90b].

One is often faced with the task of making software reliability predictions at the time a proposal is being prepared to respond to a request for proposal (RPP).

This terminology comes from the procedures used in government contracting; however, there are direct analogies in commercial contracting and perhaps less formally (but maybe they should be more formal?) for in house projects. Strictly speaking, one can not estimate the reliability of software which he or she knows nothing about, but a proper RFP will contain enough information so that the designer can liken the proposed project to a previous one. If the parallel is very close, then the only problem is to find appropriate reliability data on the previous project. More commonly, there are differences among the two projects and one must devise a technique for mapping or extrapolating the data. The following method and example is one which can be used in such circumstances.

Call the new project to be predicted 'Project A" and the prior project "Project Z." The number of hours of testing during four phases of Project Z was available: (1) prior to site integration: 2,000 hours, (2) integration at site: 4,320 hours, (3) reliability demonstration test: 700 hours, and (4) field

operation: 6,000 hours. The corresponding estimates of the number of errors removed during these four phases were; 900, 800, 14, and 33. Thus, we can calculate a failure rate over each of these four intervals of time by dividing the number of errors removed by the number of test hours.

It was postulated that this data would fit an exponentially decreasing failure rate model, and to test this hypothesis, the failure rates were plotted versus cumulative test hours on semi-log paper. The results are shown as the four horizontal bars in the accompanying figure, and the data points are shown at the center of the intervals. The solid line in Figure C.4 is fitted by eye and shows fairly good confirmation of the exponential assumption. If the new project is to be very much like the previous one the solid line can be used to estimate how reliability can be traded off versus test time in Project A.

More than likely, Project A will differ from Project Z, and in the example given, the requirements were that Project A be

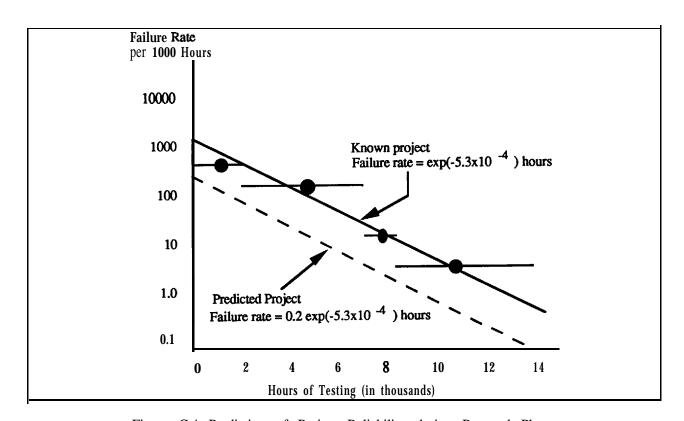


Figure C.4 Prediction of Project Reliability during Proposal Phase

much more reliable than Project Z. Project Z did not have strong software reliability and quality control focus, plan, estimation or tracking procedures (in other words, it was a normal project). The dotted line in Figure C.4 was proposed as what might be achieved if Project A had a strong reliability focus and reliability tracking. It was based on the following assumptions:

- a. A strong qualitative and quantitative reliability plan could deliver software to the integration phase with only 20% as many errors as Project Z.
- b. The errors in Project A will decline at the same rate as those did in Project Z, (even though there are fewer errors present).

Of course a complete proposal would have to include a detailed description of the reliability and quality control procedures to be used and whether the expected reductions in failure rate could meet the "goals of the estimate." Also, one would clearly feel much better about such an estimate if in addition to Project Z there were data on prior projects W, X, and Y as well and the results and circumstances of these projects corroborated the estimate.

# C.3 Select Reliability Objective

Although we are considering selection of a reliability objective here as an application in itself, it is often part of another application. In the latter case, it is part of the step of parameter determination.

There are at least three principal methods used in establishing a failure intensity objective for the software component of a system: system balance, release date, and life cycle cost minimization [STAR92]. Further discussion of these methods is given in [MUSA87, pp 194-197].

# C.4 Predict Reliability of Different Designs (Architecture)

The models given in Section C.l allow one to explore the results of a change in architecture, by examining the effect of structure on the reliability expression. One can formulate the model of the two (or more) candidate software architectures, and see how the changes effect the software reliability.

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# APPENDIX D

# RESEARCH OPPORTUNITIES

This appendix will discuss some of the known open research problems. It is not intended to be an exhaustive list of such problems.

### D. 1 Improving Parameter Estimation

Maximum likelihood estimation of parameters based on failure data taken during execution yields reasonable results, but there appears to be considerable room for improvement. Estimates are frequently biased and there is frequently considerable dispersion as well.

Joe and Reid studied the problem for an exponential binomial model (similar to but not precisely the same as the exponential Poisson model described in this document) [JOE85]. Littlewood investigated the use of adaptive prediction to compensate for estimation deficiencies [LITT86]. Further investigation into new methods of estimation may be fruitful. Modification of estimators based on measures of prediction error, an adaptation of the Littlewood approach, could be a useful approach.

This research will require failure data to be supplied through the National Repository. It will also require software reliability engineering programs in which different estimation procedures can easily be slipped in and out in modular fashion.

# **D.2 Fault Density Prediction**

Accurate means of predicting fault density are needed if we are to predict the parameters of the exponential model so that it can be used prior to program execution. At the present time, investigators have identified some of the factors that appear to affect fault density, based on a moderate number of projects. [TAKA85] found that specification change activity, average programmer skill, and thoroughness of design documentation are significant. They account for about 60% of the variation in fault density, so there are clearly other factors that are operative.

Further research is needed to address other possible factors and to verify the consistency of influence of the factors over a larger group of projects. The data required here includes number of faults identified during the life of the software, size of the software in delivered executable source lines, and measures of those factors that are likely to influence fault density. Data is needed over a wide variety of projects. The programs required are expected to be standard statistical packages.

### **D.3** Fault Exposure Ratio

The fault exposure ratio [MUSA87] is the ratio of the initial failure intensity at the start of system test to the product of the linear execution frequency and the number of inherent faults. The linear execution frequency is the average instruction execution rate divided by the object program size. It relates reliability to fault density.

Fault exposure ratio may be constant or close to it. This must be verified over a larger sample of projects. If it is not constant, then the factors that influence it need to be identified and the relationships determined.

This research requires data from a variety of projects on initial failure intensity at the start of system test, number of faults, average instruction execution rate, and object program size. It may also require information on factors that could influence the fault exposure ratio. There is no particular need for software tools.

#### **D.4 Fault Reduction Factor**

The fault reduction factor [MUSA87] is the ratio of net fault reduction to failures experienced as time of execution approaches infinity. We need to determine its value over a wide variety of projects and determine factors (if any) that affect it.

The main research requirement is data on net faults removed and failures experienced. If factors that affect fault reduction factor are identified, we need to determine their values. There is no need for software tools.

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### **D.5** Resource Usage Parameters

Information on resource usage parameters is needed on a wide variety of projects. Either they will be constant, or they will vary with factors which must be determined.

The requirements for research here are data on resource usage (failure identification effort, failure resolution effort, computer time) as a function of execution time and failures experienced. Data will also be required on the values of any variables that may affect resource usage. The program tools required will probably only be standard regression routines.

#### D.6 SRE and Unit Test

There is a good chance that software reliability estimation could be extended to unit test. There are two problems that must be addressed. First, the size of the sample of failures may be solved in grouping the failures of a number of units in some way. Second, the operational profile for the unit must be related to the system operational profile in some way or one must compensate for the difference.

The data and software tools needed for this study are not presently defined; they must be determined in the course of the study.

# D.7 Homogeneity of Failure Severity Classification

Some evidence indicates that the proportion of failures in each failure classification on a given project remains approximately constant over the life of the project.

Checking this hypothesis will require failure data from a variety of projects, with the execution time and severity classification of failure recorded.

# **D.8** Relationship Between Reliability and Problems Found During Inspection

If the inspection process happens during the coding phase of the life cycle (ex. code audits) the program is too unstable to fit a

reliability model as formulated in this document. These models attempt to fit the fault discovery process within a given environment. If the environment is rapidly changing, as would be the case during the coding phase, attempting to do reliability prediction is like trying to hit a rapidly moving target. It is not impossible but it is extremely difficult. If the code audit process is relatively stable, say over a short period of time or perhaps within a given module of the program, we might be able to fit and subsequently use our model predictions. However they would only be appropriate in a very restrictive sense. Usually the code is undergoing such rapid changes that what we attempt to model today is not the same program tomorrow!

Reliability estimation and prediction during the coding phase or earlier is an open research question. Some suggestions can however be put forth. All of these suggestions will not guarantee good results if followed. These are only recommendations based upon the experience of software developers. The first suggestion is to use past data of similar projects. One might compare the fault detection rates during the inspection process of the two similar efforts and then using the operational reliability of the past effort adjust it for the given effort. This could provide a very crude estimate of the eventual reliability of the current program.

Again extreme care needs to be exercised in extrapolating from one effort to another. Two development efforts may be similar (ex. number of lines of code, personnel, language, or intended use), however you will never have an identical development environment.

Another suggestion is to employ some of the measurements provided during the coding phase in the IEEE Standard Dictionary of Measures to Produce Reliable Software (IEEE Std 982.1-1988) and the accompanying IEEE Guide for the Use of IEEE Standard Dictionary of Measures to Produce Reliable Software (IEEE Std 982.2-1988). That effort attempted to define metrics that can be used throughout the software life cycle to measure both the

resulting product and the software process that developed it. Here the emphasis is on insuring that the effort stays on track towards its reliability goals so that when it reaches integrated testing or beyond the software models related in this document will confirm that the reliability objectives have been meet. [SIEF89] provides a basis for choosing the appropriate measures for use. Some examples of metrics that can be employed during the audit process are:

- a. Fault-days The number of days the fault has resided in the code. This could indicate problems in the process. Faults are not being discovered earlier in the life cycle when software impacts are not as great.
- b. Error Distribution For the faults discovered in the inspection process what types of errors (requirements, design, etc.) are they. This again could indicate where management needs to address changes in the software engineering process.
- c. Man-hours per major defect detected How much effort was expended in the inspection process to uncover a given fault. If this is too large (say in respect to similar development processes) the audit process may need to be modified.

The reader is encouraged to refer to those documents for additional information.

If the inspection occurs during the integration phase as part of the overall test strategy for the verification and validation of the software (V&V), the models considered in this document can be applied, assuming the software has by that time reached a relatively stable state. However again care must be considered in extrapolating the reliability predictions beyond this environment, especially to the operational phase. Generally, when code audits are performed, extensive coding reading is done. The modules are inspected each in turn with the same level of intensity. Hence modules that would not be used very often in the operational phase (or not even at all unless certain anomalies occur) are inspected at the same level as ones that occur on a regular basis. Thus faults are found at a rate that would be higher than what would normally occur within the operational phase. If this were the case, smaller Mean Time Before Failures (MTBF) would be predicted by the models than what would be observed operationally. One only needs to be aware of this danger. If one is modeling the inspection fault detection rate simply to determine whether more manpower need to be allocated or what modules need to undergo more extensive testing, then this would be an appropriate use of the models.

It is hoped that further research will provide better approaches for prediction and estimation within this important phase.

# APPENDIX E

# USING THE AIAA RECOMMENDED PRACTICE FOR SOFTWARE RELIABILITY

Section 5 outlined an eleven step **generic procedure** that can be tailored to a specific project's needs. The steps are as follows:

- 1. Identify the application under investigation
- 2. Specify the requirement
- 3. Allocate the requirement
- 4. Define failure
- 51 Characterize the operational environment
- 6. Select tests
- 7. Select model(s)
- 8. collect data
- **9.** Determine model parameters
- 10. Validate and select **best** model
- 11. Perform analysis

This document limits its scope to the period from the start of testing until system release, so while the **first** three steps are called out, they are not expanded upon in Section 5. Future research is intended to address them in detail.

Section 5 identifies considerations for each step in the procedure. The following section outlines those considerations and describes the actions taken during "Project A". It addresses each step of the generic procedure beginning with step 4 • Define Failure.

#### E. 1 Define Failure

Section 2 of this recommended practice defines failure as "The inability of a system or system component to perform a required function within specified limits." Since this is a general definition, it is recommended that a project-specific definition be negotiated by the testers, developers, and users prior to the start of test.

Prior to testing the Project A software, several meetings were held to define failure.

Developers, testers, and users decided to define failure in terms of the activity that the system would be performing (i.e., development, testing, or operations), and to categorize the failures by their severities (i.e., critical, major, or minor). The meeting results are shown in Table E.l.

Other considerations outlined in the AIAA Recommended Practice related to failure definition that require resolution are:

- Are failures counted if it is consciously decided not to seek out and remove the cause of a particular failure?
- Are duplicate failures counted each time they occur?
- Is each failure in a series that is triggered by data degradation counted individually?

Responses were no to the first question, **sometimes** to the second, and yes to the third. The rationale for answering **no** to the first question was that deciding not to correct a known fault is equivalent to changing a system requirement. Duplicate failures that were encountered using a similar test case, or during regression testing were not counted; however, if the duplicate failure was encountered using a different operational scenario the failure was counted. Answering yes to the third question simplified data collection since detailed investigation into each failure was not required prior to using the data for parameter estimation.

# E.2 Characterize the Operational Environment

The AIAA Recommended Practice defines the operational environment in terms of the system configuration, the system evolution during test, and the system's operational profile. The system configuration refers to the arrangement of the system's components. System evolution refers to changes in the design and implementation during test and the operational profile(s) refers to the relative frequency that each function of the software is executed. Each item must be considered when planning and executing the software reliability analysis.

Development	
Critical	A failure that inhibits processing in more than one area and cannot be circumvented. Additionally, a failure that requires reboot of a workstation
1 11	to correct.
Missor	A failure that inhibits processing or produces erroneous output limited to one area. Also, a failure that requires the operator to logoff then logon to restore the operation.  Anomalies that are slight and can be circumvented.
Test	Allomanes that are slight and can be circumvented.
Clitic al	Inhibits one or more applications from being tested or a failure that brings the system to a halt and cannot be circumvented. Additionally, a failure that <b>req</b> uires reboot of a workstation to correct.
Major	Inhibits an entire processor of an application from being tested or prohibits completion of a test case by blocking other test functions. Also, a failure that requires the operator to logoff then logon to restore the operation.
Minor	Failures that do not directly affect completion of a test function and are considered to have no effect in an operational environment.
Operations	
Critical	A failure that drastically reduces the usefulness of the system in support of current operations and cannot be circumvented. Additionally, a failure that requires reboot of a workstation to correct.
Major	A failure that reduces the usefulness of one or more major system functions used in current operations, and cannot conveniently be circumvented. Also, a failure that requires the operator to logoff then logon to restore the operation.
Minor	Failures that occur during a mission that are considered to have no effect or to be insignificant. <b>but are</b> to be corrected in a future release.

Table E. 1 Failure Definitions Used During Project A

### E .2.1 System Configuration

Prior to the project, the system was a basically centralized system. All processing occurred in mainframe computers. The processed data was sent to ground controllers for interpretation. Project A incorporated aspects of distributed computing into the facility by adding a Local Area Network and workstations at the users' work areas. The workstations contain software that allows the user more analysis capability while maintaining the capability from the original system.

The distributed nature of Project A raised the following two concerns for the analysis:

 Should the failure data be separated by hardware processor and should the failure rate of each component be tracked independently?

 How should the fact that different processors execute at different rates and are busy different amounts during a mission be handled?

The Project A team decided that it was only possible to track failures at the system level and not at the component level (i.e., any software component failure was a system failure), thus it was not necessary for the reliability analysis to separate the failures by component. Clearly, this simplification compromises the accuracy of the system reliability estimate by ignoring the distributed nature of the software components. However, a reliability block diagram can be constructed that would improve the accuracy of the estimate by taking into account the system architecture and functional paths.

To handle the second issue Project A collected system active test hours rather than execution time on each processor. While Section 5 recommends collecting processor execution time for completing a software reliability analysis. Unfortunately, it was not possible to measure execution time explicitly throughout the test on each processor. The system active time was defined as the time the system was processing mission simulated data. It did not include downtime due to failures, reconfigurations, or other anomalies. It does include the time the system processors spend doing Input / Output and waiting for data. It should be noted that performance measurement prior to delivery indicated that the host operates at slightly over 80% CPU busy, and multiple workstations are continuously executing during a mission. Thus, system activity is an approximation to execution time.

### E.2.2 System Evolution

Software reliability measurement models assume that the program is stable except for those changes that result from debugging. Project A evolved due to integration of parts during the test period. Three major releases were provided to the test team during the test phase. The first major release contained 308,350 source lines of code (SLOC). The second release contained an additional 486,802 SLOC for a 795,152 SLOC total. The third release added 105,722 **SLOC** for a total of 900,874 SLOC. Note that this evolution was anticipated prior to testing and all three releases contained relatively independent functionality. Furthermore, the system was stable for the final 450 active test hours.

Section 5 does not provide significant detail on how to handle this situation. It simply provides a reference. The solution for Project A was to make sure the tool they used had the capability to adjust the failure times based on the evolution of the project.

### E.2.3 System Operational Profile

Software reliability measurement models assume that the software is tested in a manner similar to operational use. The term

operational profile is used to describe the list of all operations the system can perform and the probability of occurrences of each operation.

For Project A, recorded data from previous missions and user training scenarios were used to develop the test cases. This ensured a relatively accurate operational profile for testing.

### E.3 Select Tests

Section 5 identifies two approaches to test selection. First, select tests that duplicate the operational environment of the system. Second, select tests that are more severe than the anticipated usage of the system. The second approach is intended to accelerate the test process by encountering more faults in less elapsed time.

As stated previously, Project A tested using actual data collected from previous missions. Project A also conducted a separate "stress test" of the system using simulated data that executed the software well above design limits.

### E.4 Select Models

Section 6 defines a set of model selection criteria and recommends four models as a starting point for software reliability analysis.

Project A examined each of the recommended models and determined that a special case of the Generalized Exponential model was practical for the situation. Project A experimented with several models contained in the Generalized Exponential Framework and chose the Musa Basic Model based on its goodness-of-fit and ability to handle incremental releases during test.

The other three recommended models were not practical for Project A. The fit obtained using Musa / Okumoto logarithmic Poisson model could not be validated. The Schneidewind model requires equally spaced execution intervals, which was not the case for the project since system active time was collected rather than execution or wall-clock time. Finally the Littlewood / Verrall model

implementation available to the project required time between failure data as input rather than the test interval lengths and counts that were collected.

### E. 5 Collect Data

The data collection effort must be geared toward the overall objectives of the software reliability effort. The objective for Project A was to forecast the failure rate of the software at release, and to estimate the number of software-related failures during a mission. Section 7 recommends that data collection be restricted to the data required for the specified

objectives.

Section 7 outlines a nine step data collection procedure as follows:

- 1) Establish the-objectives
- 2) Plan the data collection process
- 3) Apply tools
- 4) Provide training
- 5) Perform a trial run
- 6) Implement the plan
- 7) Monitor data collection
- 8) Use the data
- 9) Provide feedback

Date: Scheduled Time (hrs):
Effective Time (hrs):  Workstation: Host:  DSS: Simulations: other:
Test Session Rating (check one):  Excellent Good Fair - Poor -
Workstation Subsystems and Highlights:
Host Highlights:
Personnel:
Discrepancies Written: Impact Number Written
Major Minor
Itemized DR Lit: Number Impact Subsystem Description

Figure E. 1 Test Session Report Form

It also defines data primitives that should be collected for any software project. These primitives support the AIAA software reliability database and would be useful for a repository supporting future planning.

To meet Project A objectives for reliability analysis data on the number of system active hours during each test interval, the size of the software under test, and the number of recorded failures by severity during the interval were collected.

The study period consisted of 126 test sessions. At the end of each test session, a test session report form was completed by the test monitor. A sample test session report form is shown in Figure E. 1. In general, the form required the test monitor to answer several short questions; answers document the impact of all observed failures and other characteristics of the test session.

During each test session the individual testers complete a form describing each failure occurrence. The form is called a discrepancy report **(DR)**. A completed DR form contains details of the test environment and the behavior of the system when the failure occurred. At the conclusion of each test session all DR forms are delivered to the development organization for investigation and resolution.

The quality of the test session data was checked via independent inspection. Occasionally, an anomaly or contradiction arose through the inspection or subsequent analysis. If the data reporting was inconsistent across testers, the test monitor who filed the report was interviewed for clarification. For example, some testers did not fill out a DR form if a subsystem other than the one under test failed during the session. Fortunately, this data could be inferred from the summary text on the test session report form, usually in the form's "Highlights" or "Lost Time" sections. An example of such an inference is the determination of the number of failures during a test session. Since a description read "host crashed and we lost x hours while the offending subsystem's development team investigated," but no DR form was completed since the tester was "not testing the host," a failure could be inferred with rea**sonable** certainty. Data were not incorporated into the data set used for this analysis if the inference was deemed unreliable.

# E. 6 Determine Model Parameters

The AIAA recommended practice identifies three techniques to determine model parameters: 1) method of moments, 2) least squares, and 3) maximum likelihood. These are useful if the practitioner wishes to develop his / her own tool. However, the document only supplies the equations necessary to implement the maximum likelihood technique. To implement other parameter estimation techniques, the practitioner must consult sources other than this document.

To save effort on parameter determination a practitioner can select an automated tool that provides the models and estimation techniques required by your project. The **AIAA** recommended practice lists several available tools in Appendix B and lists the models each tool supports.

For project A, the SRE toolkit supplied by AT&T was used to estimate the parameters for the Musa Basic Model.

### E. 7 Validate the Model

Section 5.7 recommends validating the model "fit" on the observed data with some level of confidence using statistical tests such as **Chi**square or Kolmogorov-Smirnov. These tests are designed to detect fairly gross disagreements between the data and the fitted model.

Project A did not use either technique. Instead, they performed a visual comparison of the expected model with the actual data using the plot shown in Figure E.2. This informal heuristic procedure allowed them to feel comfortable with the model forecasts.

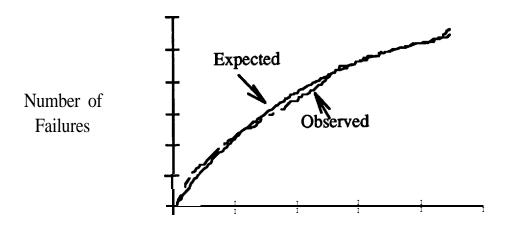
# E. 8 Perform Appropriate Analysis

Section 5.2 provides summaries of several different analysis procedures supporting

common engineering activities. Among other topics the AIAA list includes two areas of interest to Project A: (1) forecasting the current reliability, and (2) forecasting the achievement of **attaining** a reliability goal.

Figure E.3 shows the failure rate curves for each of the failure categories defined by the test team. Using these curves the current reliability can be forecast.

# Observed and Expected Failures vs Time for Musa Basic Execution Time Model



System Active Test Time (Hours)

Figure E.2 Informal Model Validation

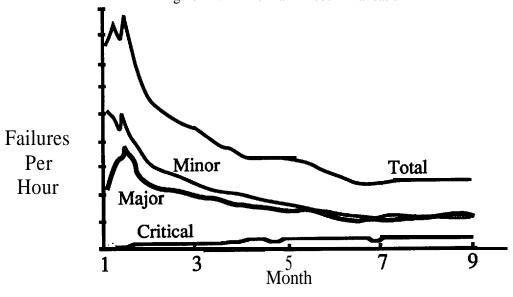


Figure E.3 Failures per Test Hour by Month

# APPENDIX F

# USING RELIABILITY MODELS FOR DEVELOPING TEST STRATEGIES

# F. 1 Allocating Test Resources

It is important for software organizations to have a strategy for testing; otherwise, test costs are likely to get out of control. Without a strategy, each module you test may be treated equally with respect to allocation of resources. You need to treat your modules unequally! That is, allocate more test time, effort and funds to the modules which have the highest predicted number of failures, F(t1,t2), during the interval t1,t2, where t1,t2 could be execution time or labor time (of testers) for a single module. In the remainder of this section, "time" means execution time. Use the convention that you make a prediction of failures at t1 for a continuous interval with end-points t1+1 and t2.

The following sections describe how a reliability model can be used to predict F(t1,t2). The test strategy is the following:

Allocate test execution time to your modules in proportion to F(t 1,t2).

Model parameters and predictions are updated based on observing the actual number of failures,  $X_{0,t1}$ , during 0,t1. This is shown in Figure F. 1, where you predict F(t1,t2), at tl during t1,t2, based on the model and  $X_{0,t1}$ . In this figure,  $t_m$  is total available test time for a single module. Note that you could have  $t2 = t_m$  (i.e., the prediction is made to the end of the test period).

Based on the updated predictions, you may want to reallocate your test resources. Of course, it could be disruptive to your organization to reallocate too frequently. So, you could predict and reallocate at major milestones (i.e., formal review of test results).

Using the Schneidewind software reliability model, and the Space Shuttle Primary Avionics Software Subsystem as an example, the process of using prediction for allocating test resources is developed. Two parameters,  $\alpha$  and  $\beta$ , which will be used in the following equations, are estimated by applying the model to  $X_{0,t1}$ . Once the parameters have been established, you can predict various quantities that will assist you in allocating test resources, as shown in the following equations:

• Number of failures during **0,t**:

$$F(t) = (\alpha/\beta)[1 - \exp(-\beta t)]$$
 (F.1)

• Using (F.1) and Figure F.1, you can predict number of failures during t 1,12:

$$F(t1,t2) = (\alpha/\beta)[1 - \exp(-\beta t2)] - X_{0,t1}$$
 (F.2)

 Also, you can predict maximum number of failures during the life (t = ∞) of the software:

$$\mathbf{F}(\infty) = (\alpha/\beta) \tag{F.3}$$

• Using (F.3), you can predict the maximum remaining number of failures at t

$$R(t) = (\alpha/\beta) \cdot X_{0,t}$$
 (F.4)

Given n modules, allocate test execution time periods  $T_i$  for each module i according to the following equation:

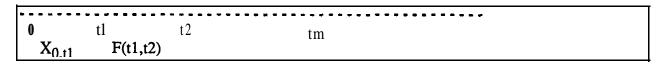


Figure F. 1 Reliability prediction time scale

$$T_{i} = \frac{F_{i}(t_{1}, t_{2})n(t_{2} - t_{1})}{\sum_{i=1}^{n} F_{i}(t_{1}, t_{2})}$$
(F.5)

In **(F.5)**, note that although predictions are made using (F.2) for a **single module**, **the** total **available** test execution time **(n)(t2 •** tl) is allocated for each module **i** across **n modules**. **You use** the same interval **0,20** for each module to estimate a and  $\beta$  and the same interval **20,30** for each module to make predictions, but from then on a variable amount of test time  $T_i$  is used depending on the predictions.

Tables F.1 and F.2 summarize the results of applying the model to the failure data for three Space Shuttle modules (operational increments). The modules are executed continuously, 24 hours per day, day after day. For illustrative purposes, each period in the test interval is assumed to be equal to 30 days. After executing the modules during 0,20, the SMERFS program was applied to the observed failure data during 0,20 to obtain estimates of a and  $\beta$ . The total number

of failures observed during **0,20** and the estimated parameters are shown in Table F.l.

Equations (F.2), (F.3), (F.4) and (F.5) were used to obtain the predictions in Table F.2 during 20,30. The prediction of F(20,30) led to the prediction of T, the allocated number of test execution time periods. The number of additional failures that were subsequently observed, as testing continued during 20,20+T, is shown as X(20,20+T). Comparing Table F.1 with Table F.2, you will see that there is the possibility of additional failures occurring in Module 1 (0.95 failures) and Module 2 (0.50 failures), based on predicted maximum number of failures  $F(\infty)$ . That is, for these modules,  $[X(0,20) + \dot{X}(20,20+T)] < F(\infty)$ . Note that the actual  $F(\infty)$  would only be known after all testing is complete and was not known at **20+T.** Thus you need additional procedures for deciding how long to test to reach a given number of remaining failures. A variant of this decision is the stopping rule (when to stop testing?). This is discussed in the following section.

Table F. 1 Observed Failures and Model Parameters

	X(0,20) Failures	α_	β
Module 1	12	1.69	0.13
Module 2	11	1.76	0.1
Module 3	10	0.68	0.0

Table F.2 Allocation of Test Resources

	F(∞) failures	F(20,30) failures	R(20) failures	T periods	X(20, 20+T) failures
Module 1 Predicted	12.95	0.695	0.952	7.6	
Actual	.00	0.000	.000		0
Nodule 2 Predicted Actual	12.5	1.32	1.5	14.4	
	13.û	1.32	2.0		1
Module 3 Predicted	10.81	0.73	0.81	8.0	
Actual	4.00	.00	4.0		1

# F. 2 Making Test Decisions

In addition to allocating test resources, you can use reliability prediction to estimate the minimum total test execution time t2 (i.e., interval 0,t2) necessary to reduce the predicted maximum number of remaining failures to R(t2). To do this, subtract equation (F. 1) from (F.3), set the result equal to R(t2), and solve for t2:

$$t2 = {\ln [(\alpha/\beta)/R(t2)]}/\beta$$
 (F.6)

where R(t2) can be established from:

$$R(t2) = (p)(\alpha/\beta)$$
 (F.7)

where p is the desired fraction (percentage) of remaining failures at t2. Substituting (F.7) into (F.6) gives:

$$t2 = {\ln [(1/p)]}/\beta$$
 (F.8)

(F.8) is plotted for Module 1 and Module 2 in Figure F.2 for various values of p

You can use (F.8) as a rule to determine when to stop testing a given module.

Using (F.8) and Figure F.2 you can produce Table F.3 which tells you the following: the total minimum test execution time t2 from time 0 to reach essentially 0 remaining failures (i.e., at p = .001 (.1%), predicted remaining failures are .01295 and .01250 for Module 1 and Module 2, respectively (see (F.7) and Table F.2)); the additional test execution time beyond 20+T shown in Table F.2; and the actual amount of test time required, starting at 0, for the "last" failure to occur (this quantity comes from the data and not from prediction). You don't know that it is necessarily the last; you only know that it

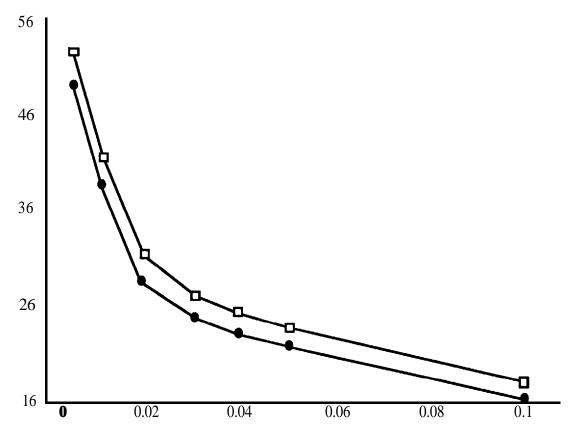


Figure F.2 Execution time needed to reach the desired fraction of remaining failures

was the "last" after 64 **periods** (1910 days) **and 44 periods** (1314days) for Module 1 and Module 2, respectively. So, **t2** = 52.9 and t2 = 49.0 periods would constitute your

stopping rule for Module 1 and Module 2, respectively. This procedure allows you to exercise control over software quality.

Table F.3 Test Time Required to Reach "0" Remaining Failures (p = .001)

	t2 periods	Additional Test Time periods	Last Failure Found periods
Module 1	52.9	45.3	64
Module 2	49.0	. 6	44

# American Institute of Aeronautics and Astronautics

The Aerospace Center 370 L'Enfant Promenade, SW Washington, DC 20024-2518

